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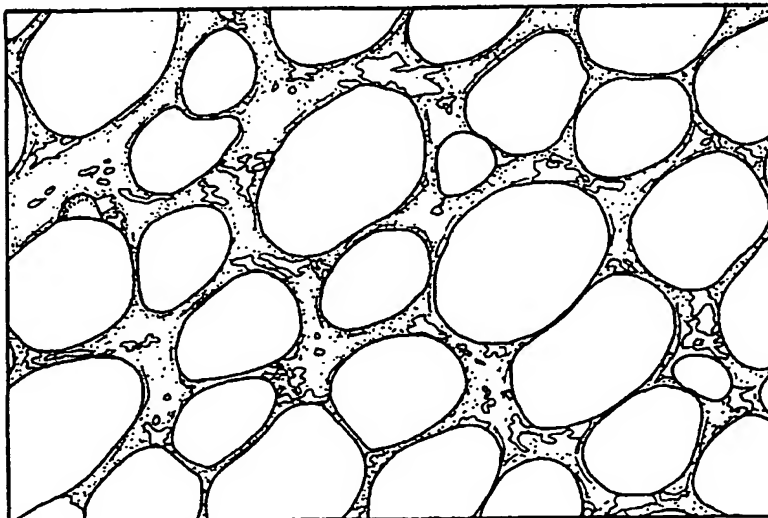
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(54) Title: PITCH-BASED CARBON FOAM AND COMPOSITES AND USES THEREOF



(57) Abstract: A thermally conductive carbon foam is provided, normally having a thermal conductivity of at least 40 W/m.K. The carbon foam usually has a specific thermal conductivity, defined as the thermal conductivity divided by the density, of at least about 75 W.cm<sup>3</sup>/m.<sup>3</sup>K.gm. The foam also has a high specific surface area, typically at least about 6,000 m<sup>2</sup>/m<sup>3</sup>. The foam is characterized by an x-ray diffraction pattern having "doublet" 100 and 101 peaks characterized by a relative peak split factor no greater than about 0.470. The foam is graphitic and exhibits substantially isotropic thermal conductivity. The foam comprises substantially ellipsoidal pores and the mean pore diameter of such pores is preferably no greater than about 340 microns. Other materials, such as phase change materials, can be impregnated in the pores in order to impart beneficial thermal properties to the foam. Heat exchange devices and evaporatively cooled heat sinks utilizing the foams are also disclosed.



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*For two-letter codes and other abbreviations, refer to the "Guidance Notes on Codes and Abbreviations" appearing at the beginning of each regular issue of the PCT Gazette.*

**PITCH-BASED CARBON FOAM AND COMPOSITES AND USES THEREOF****STATEMENT REGARDING FEDERALLY SPONSORED  
RESEARCH OR DEVELOPMENT**

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5 DE-AC05-96OR22464 between the United States Department of Energy and Lockheed  
Martin Energy Research Corporation.

**BACKGROUND OF THE INVENTION**

The present invention relates to carbon foam and composites, and more particularly to  
a process for producing them.

10 The extraordinary mechanical properties of commercial carbon fibers are due to the  
unique graphitic morphology of the extruded filaments. See Edie, D.D., "Pitch and  
Mesophase Fibers," in Carbon Fibers, Filaments and Composites, Figueiredo (editor), Kluwer  
Academic Publishers, Boston, pp. 43-72 (1990). Contemporary advanced structural  
composites exploit these properties by creating a disconnected network of graphitic filaments  
15 held together by an appropriate matrix. Carbon foam derived from a pitch precursor can be  
considered to be an interconnected network of graphitic ligaments or struts, as shown in  
Figure 1. As such interconnected networks, they represent a potential alternative as a  
reinforcement in structural composite materials.

Typical processes for producing carbon foams utilize a blowing technique to produce  
20 a foam of the pitch precursor in which the pitch is melted and passed from a high pressure  
region to a low pressure region. Thermodynamically, this produces a "Flash," thereby  
causing the low molecular weight compounds in the pitch to vaporize (the pitch boils),

resulting in a pitch foam. See Hagar, Joseph W. and Max L. Lake, "Novel Hybrid Composites Based on Carbon Foams," *Mat. Res. Soc. Symp.*, Materials Research Society, 270:29-34 (1992), Hagar, Joseph W. and Max L. Lake, "Formulation of a Mathematical Process Model Process Model for the Foaming of a Mesophase Carbon Precursor," *Mat. Res. Soc. Symp.*, Materials Research Society, 270:35-40 (1992), Gibson, L.J. and M.F. Ashby, Cellular Solids: Structures & Properties, Pergamon Press, New York (1988), Gibson, L.J., *Mat. Sci. and Eng A* 110, 1 (1989), Knippenberg and B. Lersmacher, *Phillips Tech. Rev.*, 36(4), (1976), and Bonzom, A., P. Crepau and E. J. Moutard, U.S. patent 4,276,246, (1981). Then, the pitch foam must be oxidatively stabilized by heating in air (or oxygen) for many hours, thereby, cross-linking the structure and "setting" the pitch so it does not melt during carbonization. See Hagar, Joseph W. and Max L. Lake, "Formulation of a Mathematical Process Model Process Model for the Foaming of a Mesophase Carbon Precursor," *Mat. Res. Soc. Symp.*, Materials Research Society, 270:35-40 (1992) and White, J.L., and P.M. Shaeffer, *Carbon*, 27:697 (1989). This is a time consuming step and can be an expensive step depending on the part size and equipment required. The "set" or oxidized pitch is then carbonized in an inert atmosphere to temperatures as high as 1100°C, followed by subjection to temperatures as high as 3000°C to produce a graphitic carbon foam.

Other techniques utilize a polymeric precursor, such as phenolic, urethane, or blends of these with pitch. See Hagar, Joseph W. and Max L. Lake, "Idealized Strut Geometries for Open-Celled Foams," *Mat. Res. Soc. Symp.*, Materials Research Society, 270:41-46 (1992), Aubert, J. W., (MRS Symposium Proceedings, 207:117-127 (1990), Cowlard, F.C. and J.C. Lewis, *J. of Mat. Sci.*, 2:507-512 (1967) and Noda, T., Inagaki and S. Yamada, *J. of Non-Crystalline Solids*, 1:285-302, (1969). High pressure is applied and the sample is heated. At a specified temperature, the pressure is released, thus causing the liquid to foam as volatile

compounds are released. The polymeric precursors are cured and then carbonized without a stabilization step. However, these precursors produce a "glassy" or vitreous carbon which does not exhibit graphitic structure and, thus, has low thermal conductivity and low stiffness.

See Hagar, Joseph W. and Max L. Lake, "Idealized Strut Geometries for Open-Celled

5 Foams," *Mat. Res. Soc. Symp.*, Materials Research Society, 270:41-46 (1992).

In either case, once the foam is formed, it is then bonded in a separate step to the facesheet used in the composite. This can be an expensive step in the utilization of the foam.

The process of this invention overcomes these limitations, by not requiring a "blowing" or "pressure release" technique to produce the foam. Furthermore, an oxidation  
10 stabilization step is not required, as in other methods used to produce pitch-based carbon foams. This process is less time consuming, and therefore, will be lower in cost and easier to fabricate. Moreover, the foam can be produced with an integrated sheet of high thermal conductivity carbon on the surface of the foam, thereby producing a carbon foam with a smooth sheet on the surface to improve heat transfer.

15 The present invention further relates to a thermally-conductive foam material derived from carbonaceous precursor, and more particularly to a thermally conductive, pitch-derived carbon foam having high thermal conductivity and heat exchanging properties.

The removal of unwanted heat is a frequently encountered problem. Conventional solutions include cooling fans, ice packs and refrigeration systems. In the latter, a working  
20 fluid is compressed (condensed) and pumped into an expansive chamber or pipe system where it evaporates, pulling heat from the atmosphere to satisfy its needed latent heat of vaporization, and thus cooling the surrounding environment. Air blown through the heat exchanger may be cooled and circulated to cool larger volumes such as in domestic and automotive air conditioning systems.

Active cooling (refrigeration) typically requires complex equipment including pumps, valves, compressors, etc. Many refrigeration systems require the use of CFCs (Freon), which is considered hazardous or environmentally unfriendly. An evaporative cooling system with a high thermal conductivity medium would offer a simpler, lower cost alternative. There is a  
5 need for portable coolers which are lightweight and inexpensive so as to be deployed in the field or in third world countries.

The thermally conductive carbon foam of this invention overcomes the limitations of the prior art.

In addition, the present invention relates to porous carbon foam filled with phase  
10 change materials and encased to form a heat sink product, and more particularly to a process for producing them.

There are currently many applications that require the storage of large quantities of heat for either cooling or heating an object. Typically these applications produce heat so rapidly that normal dissipation through cooling fins, natural convection, or radiation cannot  
15 dissipate the heat quickly enough and, thus, the object over-heats. To alleviate this problem, a material with a large specific heat capacity, such as a heat sink, is placed in contact with the object as it heats. During the heating process, heat is transferred to the heat sink from the hot object, and as the heat sink's temperature rises, it "stores" the heat more rapidly than can be dissipated to the environment through convection. Unfortunately, as the temperature of the  
20 heat sink rises the heat flux from the hot object decreases, due to a smaller temperature difference between the two objects. Therefore, although this method of energy storage can absorb large quantities of heat in some applications, it is not sufficient for all applications.

Another method of absorbing heat is through a change of phase of the material, rather than a change in temperature. Typically, the phase transformation of a material absorbs two

orders of magnitude greater thermal energy than the heat capacity of the material. For example, the vaporization of 1 gram of water at 100°C absorbs 2,439 Joules of energy, whereas changing the temperature of water from 99°C to 100°C only absorbs 4.21 Joules of energy. In other words, raising the temperature of 579 grams of water from 99°C to 100°C  
5 absorbs the same amount of heat as evaporating 1 gram of water at 100°C. The same trend is found at the melting point of the material. This phenomenon has been utilized in some applications to either absorb or evolve tremendous amounts of energy in situations where heat sinks will not work.

Although a solid block of phase change material has a very large theoretical capacity  
10 to absorb heat, the process is not a rapid one because of the difficulties of heat transfer and thus it cannot be utilized in certain applications. However, the utilization of the high thermal conductivity foam will overcome the shortcomings described above. If the high conductivity foam is filled with the phase change material, the process can become very rapid. Because of the extremely high conductivity in the struts of the foam, as heat contacts the surface of the  
15 foam, it is rapidly transmitted throughout the foam to a very large surface area of the phase change material. Thus, heat is very quickly distributed throughout the phase change material, allowing it to absorb or emit thermal energy extremely quickly without changing temperature, thus keeping the driving force for heat transfer at its maximum.

Heat sinks have been utilized in the aerospace community to absorb energy in  
20 applications such as missiles and aircraft where rapid heat generation is found. A material that has a high heat of melting is encased in a graphite or metallic case, typically aluminum, and placed in contact with the object creating the heat. Since most phase change materials have a low thermal conductivity, the rate of heat transfer through the material is limited, but this is offset by the high energy absorbing capability of the phase change. As heat is

transmitted through the metallic or graphite case to the phase change material, the phase change material closest to the heat source begins to melt. Since the temperature of the phase change material does not change until all the material melts, the flux from the heat source to the phase change material remains relatively constant. However, as the heat continues to melt  
5 more phase change material, more liquid is formed. Unfortunately, the liquid has a much lower thermal conductivity, thus hampering heat flow further. In fact, the overall low thermal conductivity of the solid and liquid phase change materials limits the rate of heat absorption and, thus, reduces the efficiency of the system.

#### SUMMARY OF THE INVENTION

10 The general object of the present invention is to provide carbon foam and a composite from a mesophase or isotropic pitch such as synthetic, petroleum or coal-tar based pitch.

Another object is to provide a carbon foam and a composite from pitch which does not require an oxidative stabilization step.

These and other objectives are accomplished by a method of producing carbon foam  
15 wherein an appropriate mold shape is selected and preferably an appropriate mold release agent is applied to walls of the mold. Pitch is introduced to an appropriate level in the mold, and the mold is purged of air such as by applying a vacuum. Alternatively, an inert fluid could be employed. The pitch is heated to a temperature sufficient to coalesce the pitch into a liquid which preferably is of about 50°C to about 100°C above the softening point of the  
20 pitch. The vacuum is released and an inert fluid applied at a static pressure up to about 1000 psi. The pitch is heated to a temperature sufficient to cause gases to evolve and foam the pitch. The pitch is further heated to a temperature sufficient to coke the pitch and the pitch is cooled to room temperature with a simultaneous and gradual release of pressure.



In another aspect, the previously described steps are employed in a mold composed of a material such that the molten pitch does not wet.

In yet another aspect, the objectives are accomplished by the carbon foam product produced by the methods disclosed herein including a foam product with a smooth integral  
5 facesheet.

In still another aspect a carbon foam composite product is produced by adhering facesheets to a carbon foam produced by the process of this invention.

Another object of the present invention is to provide a thermally conductive carbon foam.

10 Yet another object is to provide a method of producing a cooling effect utilizing a thermally conductive carbon foam.

Still another object is to provide a heat exchanging device employing a carbon foam core.

These and other objectives are accomplished in one embodiment by a thermally  
15 conductive, pitch-derived carbon foam.

In one aspect the foam has an open cell ligament composition.

In another embodiment, the objectives are accomplished by a method of producing a cooling effect wherein a thermally conductive, pitch-derived carbon foam is selected. The foam is contacted with an evaporating liquid, and an evaporation of the evaporating liquid is  
20 effected.

In still another embodiment, the objectives are accomplished by a heat exchanging device having a thermally conductive, pitch-derived carbon foam core. A fluid impermeable coating covers a portion of the foam core and exposes a portion. The exposed portion provides access and egress for an evaporating liquid.

In another aspect, there are upper and lower reservoirs in fluid communication with a core and a pumping device in fluid communication with the upper and lower reservoir adapted to deliver the evaporating liquid from the lower reservoir to the upper reservoir.

In still another aspect, the carbon foam is positioned in separate columns to provide a cold storage container with spacing between the columns.

In yet another aspect, relative motion between the foam and heat transfer fluid is developed in the presence or absence of an evaporative liquid by moving the foam, thereby accelerating evaporation and increasing the cooling effect.

Still another object of the present invention is the production of encased high thermal conductivity porous carbon foam filled with a phase change material wherein tremendous amounts of thermal energy are stored and emitted very rapidly. The porous foam that has been filled with a phase change material (PCM) will conduct heat to the phase change material such that the temperature of the phase change material will remain close to the operating temperature of the device. As heat is added to the surface, from a heat source such as a computer chip, friction due to re-entry through the atmosphere, or radiation such as sunlight, it is transmitted rapidly and uniformly throughout the foam and then to the phase change material. As the material changes phase, it absorbs orders of magnitude more energy than non-PCM material due to transfer of the latent heat of fusion or vaporization. Conversely, the filled foam can be utilized to emit energy rapidly when placed in contact with a cold object.

Non-limiting embodiments disclosed herein are a device to rapidly thaw frozen foods or freeze thawed foods, a design to prevent overheating of satellites or store thermal energy as they experience cyclic heating during orbit, and a design to cool leading edges during hypersonic flight or re-entry from space.

## BRIEF DESCRIPTION OF THE DRAWINGS

Fig. 1 is a micrograph illustrating typical carbon foam with interconnected carbon ligaments and open porosity.

5 Figs. 2-6 are micrographs of pitch-derived carbon foam graphitized at 2500°C and at various magnifications.

Fig. 7 is a drawing corresponding to a SEM micrograph (shown in Figure 27) of the foam produced by the process of this invention.

Fig. 8 is a chart illustrating cumulative intrusion volume versus pore diameter.

Fig. 9 is a chart illustrating log differential intrusion volume versus pore diameter.

10 Fig. 10 is a graph illustrating the temperatures at which volatiles are given off from raw pitch.

Fig. 11 is an X-ray analysis of the graphitized foam produced by the process of this invention.

15 Figs. 12 A-C are photographs illustrating foam produced with aluminum crucibles and the smooth structure or face sheet that develops.

Fig. 13A is a schematic view illustrating the production of a carbon foam composite made in accordance with this invention.

Fig. 13B is a perspective view of the carbon foam composite of this invention.

5 Figs. 14-16 are charts plotting temperature/time of the carbon foam resulting from the evaporation of a working fluid according to this invention.

Fig. 17 is a diagrammatic view illustrating one embodiment employing the carbon foam of this invention.

Fig. 18-21 are diagrammatic views illustrating other embodiments employing the carbon foam of this invention.

10 Fig. 22 is section cut of a heat sink device for thawing food using acetic acid as the phase change material.

Fig. 23 is a section cut of a heat sink to prevent overheating of satellites during cyclic orbits.

Fig. 24 is a section cut of a heat sink used on the leading edge of a shuttle orbiter.

15 Fig. 25 is a chart plotting the thermal conductivity as a function of density for ARA24 mesophase derived graphite foam graphitized at 4°C/min and 10°C/min.

Fig. 26 is a chart plotting the thermal conductivity as a function of density for Conoco mesophase derived graphite foam graphitized at 10°C/min.

Fig. 27 is a photograph taken by SEM imaging of a sample of the carbon foam of the invention.

5            Fig. 28 is a photograph taken by SEM imaging of a sample of the carbon foam of the invention illustrating the open interconnects between cells and showing how the interconnect diameter is about half that of the cell diameter, typically on the order of 40% to 60% of the cell diameter.

#### DETAILED DESCRIPTION OF THE INVENTION

10            In order to illustrate the carbon foam product and composite of this invention, the following Examples I-XIX are set forth. They are not intended to limit the invention in any way.

##### EXAMPLE I:

            Pitch powder, granules, or pellets are placed in a mold with the desired final shape of  
15           the foam. These pitch materials can be solvated if desired. In this Example Mitsubishi ARA-24 mesophase pitch was utilized. A proper mold release agent or film is applied to the sides of the mold to allow removal of the part. In this case, Boron Nitride spray and Dry Graphite Lubricant were separately used as a mold release agent. If the mold is made from pure aluminum, no mold release agent is necessary since the molten pitch does not wet the  
20           aluminum and, thus, will not stick to the mold. Similar mold materials may be found that the pitch does not wet and, thus, they will not need mold release. The sample is evacuated to less

than 1 torr and then heated to a temperature approximately 50 to 100°C above the softening point. In this case where Mitsubishi ARA24 mesophase pitch was used, 300°C was sufficient. At this point, the vacuum is released to a nitrogen blanket and then a pressure of up to 1000 psi is applied. The temperature of the system is then raised to 800°C, or a

5 temperature sufficient to coke the pitch which is about

500°C to about 1000°C. This is performed at a rate of no greater than about 5°C/min. and preferably at about 2°C/min. The temperature is held for at least 15 minutes to achieve an assured soak and then the furnace power is turned off and cooled to room temperature.

Preferably the foam was cooled at a rate of approximately 1.5°C/min. with release of pressure

10 at a rate of approximately 2 psi/min. Final foam temperatures for three product runs were 500°C, 630°C and 800°C. During the cooling cycle, pressure is released gradually to atmospheric conditions. The foam was then heat treated to 1050°C (carbonized) under a nitrogen blanket and then heat treated in separate runs to 2500°C and 2800°C (graphitized) in Argon.

15 Carbon foam produced with this technique was examined with photomicrography, scanning electron microscopy (SEM), X-ray analysis, and mercury porosimetry. As can be seen in the Figures 2-7, the interference patterns highlighting the isochromatic regions under cross-polarized light indicate that the struts of the foam are completely graphitic. That is, all of the pitch was converted to graphite and aligned along the axis of the struts. These struts

20 are also similar in size and are interconnected throughout the foam. This would indicate that the foam would have high stiffness and good strength. As seen in Fig. 7 by the SEM micrograph of the foam, the foam is open cellular meaning that the porosity is not closed. Figures 8 and 9 are results of the mercury porosimetry tests. These tests indicate that the pore sizes are in the range of 90-200 microns.

A thermogravimetric study of the raw pitch was performed to determine the temperature at which the volatiles are evolved. As can be seen in Figure 10, the pitch loses nearly 20% of its mass fairly rapidly in the temperature range between about 420°C and about 480°C. Although this was performed at atmospheric pressure, the addition of 1000 psi pressure will not shift this effect significantly. Therefore, while the pressure is at 1000 psi, gases rapidly evolved during heating through the temperature range of 420°C to 480°C. The gases produce a foaming effect (like boiling) on the molten pitch. As the temperature is increased further to temperatures ranging from 500°C to 1000°C (depending on the specific pitch), the foamed pitch becomes coked (or rigid), thus producing a solid foam derived from pitch. Hence, the foaming has occurred before the release of pressure and, therefore, this process is very different from previous art.

Samples from the foam were machined into specimens for measuring the thermal conductivity. The bulk thermal conductivity ranged from 58 W/m•K to 106 W/m•K. The average density of the samples was 0.53 g/cm<sup>3</sup>. When weight is taken into account, the specific thermal conductivity of the pitch derived foam is over 4 times greater than that of copper. Further derivations can be utilized to estimate the thermal conductivity of the struts themselves to be nearly 700 W/m•K. This is comparable to high thermal conductivity carbon fibers produced from this same ARA24 mesophase pitch.

X-ray analysis of the foam was performed to determine the crystalline structure of the material. The x-ray results are shown in Figure 11. From this data, the graphene layer spacing ( $d_{002}$ ) was determined to be 0.336 nm. The coherence length ( $L_a$ , 1010) was determined to be 203.3 nm and the stacking height was determined to be 442.3 nm.

The compression strength of the samples were measured to be 3.4 MPa and the compression modulus was measured to be 73.4 MPa. The foam sample was easily machined and could be handled readily without fear of damage, indicating a good strength.

It is important to note that when this pitch is heated in a similar manner, but under only atmospheric pressure, the pitch foams dramatically more than when under pressure. In fact, the resulting foam is so fragile that it could not even be handled to perform tests. Molding under pressure serves to limit the growth of the cells and produces a usable material.

#### EXAMPLE II:

An alternative to the method of Example I is to utilize a mold made from aluminum. In this case two molds were used, an aluminum weighing dish and a sectioned soda can. The same process as set forth in Example I is employed except that the final coking temperature was only 630°C, so as to prevent the aluminum from melting.

Figures 12 A-C illustrate the ability to utilize complex shaped molds for producing complex shaped foam. In one case, shown in Fig. 12 A, the top of a soda can was removed and the remaining can used as a mold. No release agent was utilized. Note that the shape of the resulting part conforms to the shape of the soda can, even after graphitization to 2800°C. This demonstrates the dimensional stability of the foam and the ability to produce near net shaped parts.

In the second case, as shown in Figs. 12 B and C employing an aluminum weight dish, a very smooth surface was formed on the surface contacting the aluminum. This is directly attributable to the fact that the molten pitch does not wet the surface of the aluminum. This would allow one to produce complex shaped parts with smooth surfaces so as to improve contact area for bonding or improving heat transfer. This smooth surface will act as a face sheet and, thus, a foam-core composite can be fabricated in-situ with the fabrication of



the face sheet. Since it is fabricated together as an integral material, no interface joints result and thermal stresses will be less, resulting in a stronger material.

The following examples illustrate the production of a composite material employing the foam of this invention.

5      EXAMPLE III:

Pitch derived carbon foam was produced with the method described in Example I. Referring to Fig. 13A the carbon foam 10 was then machined into a block 2"x2"x1/2". Two pieces 12 and 14 of a prepeg comprised of Hercules AS4 carbon fibers and ICI Fibrice Polyetheretherketone thermoplastic resin also of 2"x2"x1/2" size were placed on the top and  
10      bottom of the foam sample, and all was placed in a matched graphite mold 16 for compression by graphite plunger 18. The composite sample was heated under an applied pressure of 100 psi to a temperature of 380°C at a rate of 5°C/min. The composite was then heated under a pressure of 100 psi to a temperature of 650°C. The foam core sandwich panel generally 20 was then removed from the mold and carbonized under nitrogen to 1050°C and  
15      then graphitized to 2800°C, resulting in a foam with carbon-carbon facesheets bonded to the surface. The composite generally 30 is shown in Figure 13B.

EXAMPLE IV:

Pitch derived carbon foam was produced with the method described in Example I. It was then machined into a block 2"x2"x1/2". Two pieces of carbon-carbon material,  
20      2"x2"x1/2", were coated lightly with a mixture of 50% ethanol, 50% phenolic Durez® Resin available from Occidental Chemical Co. The foam block and carbon-carbon material were positioned together and placed in a mold as indicated in Example III. The sample was heated to a temperature of 150°C at a rate of 5°C/min and soaked at temperature for 14 hours. The sample was then carbonized under nitrogen to 1050°C and then graphitized to 2800°C,

resulting in a foam with carbon-carbon facesheets bonded to the surface. This is also shown generally at 30 in Figure 13B.

#### EXAMPLE V:

Pitch derived carbon foam was produced with the method described in Example I.

- 5 The foam sample was then densified with carbon by the method of chemical vapor infiltration for 100 hours. The density increased to  $1.4 \text{ g/cm}^3$ , the flexural strength was 19.5 MPa and the flexural modulus was 2300 MPa. The thermal conductivity of the raw foam was  $58 \text{ W/m}\cdot\text{K}$  and the thermal conductivity of the densified foam was  $94 \text{ W/m}\cdot\text{K}$ .

#### EXAMPLE VI:

- 10 Pitch derived carbon foam was produced with the method described in Example I. The foam sample was then densified with epoxy by the method of vacuum impregnation. The epoxy was cured at  $150^\circ\text{C}$  for 5 hours. The density increased to  $1.37 \text{ g/cm}^3$  and the flexural strength was measured to be 19.3 MPa.

- 15 It is obvious that other materials, such as metals, ceramics, plastics, or fiber reinforced plastics could be bonded to the surface of the foam of this invention to produce a foam core composite material with acceptable properties. It is also obvious that ceramics, or glass, or other materials could be impregnated into the foam for densification.

Based on the data taken to date from the carbon foam material, several observations can be made and the important features of the invention are:

- 20 1. Pitch-based carbon foam can be produced without an oxidative stabilization step, thus saving time and costs.
2. High graphitic alignment in the struts of the foam is achieved upon graphitization to  $2500^\circ\text{C}$ , and thus high thermal conductivity and stiffness will be exhibited by the foam, making them suitable as a core material for thermal applications.

3. High compressive strengths should be achieved with mesophase pitch-based carbon foams, making them suitable as a core material for structural applications.
4. Foam core composites can be fabricated at the same time as the foam is generated, thus saving time and costs.
5. Rigid monolithic preforms can be made with significant open porosity suitable for densification by the Chemical Vapor Infiltration method of ceramic and carbon infiltrants.
6. Rigid monolithic preforms can be made with significant open porosity suitable for activation, producing a monolithic activated carbon.
7. It is obvious that by varying the pressure applied, the size of the bubbles formed during the foaming will change and, thus, the density, strength, and other properties can be affected.

The following alternative procedures and products can also be effected by the process of this invention:

1. Fabrication of preforms with complex shapes for densification by CVI or Melt Impregnation.
2. Activated carbon monoliths.
3. Optical absorbent.
4. Low density heating elements.
5. Firewall Material.
6. Low secondary electron emission targets for high-energy physics applications.

It will thus be seen that the present invention provides for the manufacture of pitch-based carbon foam for structural and thermal composites. The process involves the fabrication of a graphitic foam from a mesophase or isotropic pitch which can be synthetic,

petroleum, or coal-tar based. A blend of these pitches can also be employed. The simplified process utilizes a high pressure high temperature furnace and thereby, does not require an oxidative stabilization step. The foam has a relatively uniform distribution of pore sizes ( $\approx 100$  microns), very little closed porosity, and density of approximately  $0.53 \text{ g/cm}^3$ . The mesophase pitch is stretched along the struts of the foam structure and thereby produces a highly aligned graphitic structure in the struts. These struts will exhibit thermal conductivities and stiffness similar to the very expensive high performance carbon fibers (such as P-120 and K1100). Thus, the foam will exhibit high stiffness and thermal conductivity at a very low density ( $\approx 0.5 \text{ g/cc}$ ). This foam can be formed in place as a core material for high temperature sandwich panels for both thermal and structural applications, thus reducing fabrication time.

By utilizing an isotropic pitch, the resulting foam can be easily activated to produce a high surface area activated carbon. The activated carbon foam will not experience the problems associated with granules such as attrition, channeling, and large pressure drops.

The high thermal conductivity carbon foam of the invention may be utilized to provide an evaporatively cooled heat sink or heat exchanger. The carbon foam, as derived from mesophase pitch and as depicted in Figs. 2-7, has an open structure which allows free access to a working fluid to the cell walls/ligaments. When the working fluid contacts the cell surface it evaporates, and the latent heat of vaporization causes cooling of the carbon foam. The extent of cooling depends upon the working fluid and the ambient conditions (temperature and pressure). The heat sink/exchanger temperature has been shown to fall to less than  $223\text{K}$  ( $-50^\circ\text{C}$ ) using acetone as the working fluid at a pressure of  $1200$  microns Hg ( $1.2$  torr), and  $0.5^\circ\text{C}$  using acetone as the working fluid at ambient temperature and pressure. Forced air flow over the carbon foam increases the temperature drop in excess of that

observed under ambient conditions. The heat sink/exchanger described herein finds applications in heat removal systems such as personal/body cooling suits, portable refrigeration systems or coolers, and air conditioning systems (household and automotive).

The following Examples demonstrate the evaporative cooling effect on the previously described carbon foam when contacted with different working fluids as represented by acetone, ethanol and water. These Examples are not intended to limit the invention in any way. The foamed carbon was doused or partially immersed in the working fluid. Upon removal from the working fluid, and as indicated in Examples VII-X, the foam sample was placed in a vacuum furnace with a thermocouple penetrating the foam sample. The foam temperature was monitored as a function of time and pressure (vacuum). The ambient laboratory temperature was approximately 21°C.

EXAMPLE VII: *Acetone*

	Time(minutes)	Pressure(Torr)	Temperature(°C)
	0	740	13.5
15	1	29	-37.5
	2	29	-46.7
	3	1.2	-51.8
	4	1.2	-53.4

When the sample was removed from the vacuum furnace it was noted that ice had formed, presumably from moisture condensed from the furnace atmosphere, or desorbed from the foam.

EXAMPLE VIII: *Ethanol*

	Time(minutes)	Pressure(Torr)	Temperature(°C)
	0	740	20.5
25	1	29	5.3
	2	29	-14.7
	3	1.2	-21.7

Time(minutes)	Pressure(Torr)	Temperature(°C)
4	1.2	-25.1
5	1.1	-26.8
6	1.0	-28.6

5      EXAMPLE IX: *Water*

	Time(minutes)	Pressure(Torr)	Temperature(°C)
	0	740	20.5
	1	29	16.4
	2	29	16.5
10	3	29	16.6
	4	29	14.6
	5	29	12.9
	6	29	10.5
	7	29	2.6
15	8	29	-1.5
	9	29	-5.5

In the instance of Example IX the sample was immersed in water in vacuum to ensure that the foam was saturated. This probably allowed an excess of water to penetrate the sample and reduced the exposed foam surface area available for evaporation. Moreover, the resultant high water partial pressure in the furnace made it impossible to attain good vacuum in a reasonable time. Consequently, the experiment was repeated in Example X, but with substantially less water applied to the foam.

EXAMPLE X: *Water* (Repeat)

	Time(minutes)	Pressure(Torr)	Temperature(°C)
25	0	740	19.9
	1	29	14.5
	2	29	0.3
	3	29	-5.5

In this case, sub-zero temperatures were attained in a much shorter time than for Example IX.

The data for Examples VII, VIII and X are plotted in Fig. 14. The lowest temperature observed ( $-53.4^{\circ}$ ) was attained in 4 minutes using acetone as the working fluid. Temperatures of  $-24.1^{\circ}\text{C}$  and  $-5.5^{\circ}\text{C}$  were attained over the same time period when the working fluid was ethanol and water, respectively.

5 A further series of tests as set forth in Examples XI-XIII were performed to show the effect of evaporative cooling at atmospheric pressure and temperature. The foamed carbon sample was placed in a petri dish. A thermocouple was located in a hole machined into the foam. The carbon foam was doused with the working fluid until the bottom of the petri dish was completely covered with the working fluid. The resultant foam temperature was then  
10 noted as a function of time.

Example XI: *Acetone*

	Time(minutes)	Temperature( $^{\circ}\text{C}$ )	Time(minutes)	Temperature( $^{\circ}\text{C}$ )
	0	21.7	19	3.4
	1	15.7	20	3.2
15	2	13.6	21	3.0
	3	11.5	22	2.9
	4	10.3	23	2.7
	5	8.9	24	2.6
	6	8.0	25	2.4
20	7	7.3	26	2.3
	8	6.6	27	2.1
	9	6.1	28	2.0
	10	5.7	29	1.8
	11	5.3	30	1.6
25	12	4.9	31	1.4
	13	4.5	32	1.3
	14	4.3	33	1.1
	15	4.1	34	1.0
	16	3.9	35	0.8
30	17	3.7	36	0.7
	18	3.5	37	0.6
			38	0.5

After 38 minutes there was no acetone visible in the petri dish or under the carbon foam sample. The sample was placed in an air circulating oven at 60°C to dry it and then allowed to cool to ambient temperature.

EXAMPLE XII: *Ethanol*

5	Time(minutes)	Temperature(°C)	Time(minutes)	Temperature(°C)
	0	21.6	19	15.4
	1	20.3	20	15.3
	2	19.6	21	15.1
	3	19.0	22	15.0
10	4	18.6	23	15.0
	5	18.1	24	14.9
	6	17.8	25	14.8
	7	17.4	26	14.8
	8	17.1	27	14.8
15	9	16.9	28	14.7
	10	16.7	29	14.7
	11	16.5	30	14.6
	12	16.3	31	14.6
	13	16.2	32	14.6
20	14	16.0	33	14.5
	15	15.8	34	14.5
	16	15.7	35	14.4
	17	15.6	36	14.4
	18	15.5	37	14.4
25			38	14.3

After 38 minutes there was a significant amount of ethanol visible in the bottom of the petri dish. The sample was placed in an air circulating oven at 60°C to dry it and then allowed to cool to ambient temperature.

EXAMPLE XIII: *Water*

30	Time(minutes)	Temperature(°C)	Time(minutes)	Temperature(°C)
	0	20.9	19	19.3
	1	20.3	20	19.3
	2	20.2	21	19.3



	Time(minutes)	Temperature(°C)	Time(minutes)	Temperature(°C)
	3	20.1	22	19.2
	4	19.9	23	19.2
	5	19.8	24	19.1
5	6	19.7	25	19.1
	7	19.6	26	19.1
	8	19.5	27	19.1
	9	19.5	28	19.1
	10	19.5	29	19.1
10	11	19.5	30	19.0
	12	19.5	31	19.0
	13*	19.5	32	19.0
	14	19.4	33	19.0
	15	19.4	34	18.9
15	16	19.3	35	18.9
	17	19.3	36	18.9
	18	19.3	37	18.9
			38	18.9

\* Additional water squirted over carbon foam sample.

20 After 38 minutes there was a significant amount of water visible in the bottom of the petri dish. The temperature of the carbon foam plotted as a function of time is shown in Fig. 15. The minimum temperatures are higher for all three working fluids than in the previous Examples where evaporation occurred under vacuum. Moreover, the rate of temperature decrease was much smaller for all three of the working fluid under ambient conditions. The

25 lowest temperature reached (0.5°C) was attained in 38 minutes with acetone as the working fluid. The lowest temperatures attained over similar time periods were 14.3°C and 18.9°C for ethanol and water, respectively.

A third series of tests were conducted to determine the effect on foam temperature of enhanced air flow during the evaporative cooling process. The procedure set forth in the

30 previous Example was followed, except that in this series of experiments a fan (rotary, electric motor driven domestic cooling type) was used to blow ambient air across the foam and petri dish.

EXAMPLE XIV: *Acetone* with forced air flow

	Time(minutes)	Temperature(°C)	Time(minutes)	Temperature(°C)
	0	21.5	4	-2.8
	1	5.2	5	-3.2
5	2	-0.9	6	-3.5
	3	-2.9	7	-3.7

Petri dish was frequently replenished with additional acetone.

EXAMPLE XV: *Ethanol* with forced air flow

	Time(minutes)	Temperature(°C)	Time(minutes)	Temperature(°C)
	0	21.1	6	9.1
10	1	14.6	7	8.9
	2	11.5	8	8.7
	3	10.8	9	8.8
	4	9.7	10	8.9
15	5	9.3		

Ethanol in Petri dish replenished once.

EXAMPLE XVI: *Water* with forced air flow

	Time(minutes)	Temperature(°C)	Time(minutes)	Temperature(°C)
	0	21.1	8	15.0
20	1	18.7	9	14.9
	2	17.1	10	14.8
	3	16.5	11	14.8
	4	15.9	12	14.7
	5	15.6	13	14.7
25	6	15.3	14	14.7
	7	15.1	15	14.6

The data from Examples XIV, XV, and XVI are plotted in Fig. 16. An enhanced cooling effect is obtained when air is forced over the evaporating working fluid/carbon foam.

Table I below summarizes the temperature drops (differences) attained for each working fluid under the three sets of conditions employed.

Table I. Summary of temperature drop data for the three conditions examined here.

Working Fluid	Temperature Drop, °C		
	Vacuum	Ambient Pressure and Forced Air Flow	Ambient Pressure
Acetone	66.9	25.2	21.2
Ethanol	49.1	12.4	7.3
Water	26	6.5	2.0

5

The temperature drops recorded in the above Table I for a vacuum represent extreme conditions. Lower temperature drops would be attained if intermediate vacuum pressures were used, as indicated by the ambient data. Forced air flow enhanced the cooling effect due to evaporation because the partial pressure of the evaporated solvent over the foam was reduced, and the saturated air was being constantly purged with fresh (unsaturated) air.

10

These data clearly demonstrate that the carbon foam of this invention readily attains very low temperatures, due to the evaporative cooling effect of the working fluid, which can be used for the removal of unwanted heat. The three example working fluids employed in the Examples were selected because of their availability. An ideal working fluid would have a high latent heat of vaporization, a vaporization temperature close to ambient, be non-toxic and environmentally acceptable.

15

The foam material of this invention attains low temperatures for several reasons: (i) It is an efficient heat transfer medium because of its excellent thermal conductivity and large surface area; (ii) The working fluid has a high latent heat of vaporization and a low temperature (close to room temperature); (iii) The ambient pressure is low (i.e., a vacuum) causing rapid evaporation from the carbon foam surface.

20

The following are descriptions of preferred embodiments of heat removal systems for different applications that take advantage of the low temperature attained in the foam of this invention through evaporative cooling:

An evaporatively cooled heat sink or air conditioner for home or automobile is illustrated in Fig. 17 generally at 110. A working fluid is pumped from a reservoir 112 to a header tank 114 via pump 116 and lines 15 and 17. It drains through the carbon foam 118 of this invention which is encased in a impermeable coating or skin 120. The downward flow of fluid through the foam 118 occurs under the influence of gravity or a pressure differential created by the pump 116. Evaporation of the working fluid from the carbon foam surface causes cooling of the carbon foam 118. A vacuum in the reservoir 112 created by pump 116 enhances evaporative cooling from the foam 118 and increase the temperature drop, as demonstrated in the previous Examples VII-XVI. A fan with a motor 22 and duct 24 directs a separate air stream (at ambient temperature) from the air used for evaporation through penetrations 26 in the coating or casing 120 and foam core 118 where the air gives up excess heat to the cooled foam core 118. The air therefore exits the foam core 118 at below ambient temperature where it may be ducted to cool inhabited space. Condensers or cold traps 28 may be required to condense vapor exiting the foam core 118. The condensed working fluid is returned to the header tank 114.

Alternatively, instead of air another cooling fluid, such as water, ethylene glycol, helium or nitrogen could be used to remove heat from critical components, such as electronics or chemical/medicines in cold storage, or internal combustion engines.

Fig. 18 shows an evaporatively cooled cold box generally 130. An encapsulated carbon foam core 32 surrounds a series of open cavities 36 into which items to be cooled are placed. The encapsulating skin 38 also provides enclosed cavities 40 and 42 above and

below the foam core 32. The working fluid is poured into the top closed cavity 40 such as through opening 44 and drains through the foam 32. Vents 46 are additionally located in the top cavity allowing the working fluid to evaporate to the atmosphere. Evaporation of the working fluid from the carbon foam 32 surface reduces the foam's temperature. Heat for additional working fluid evaporation is extracted from the open cavities 36, thus reducing the temperature within the cavities. The entire cold box is wrapped or clad in the thermal insulation and a thermally insulated lid 48 seals the open (cold storage) cavities. A fan could be fitted to the insulated cold box 130 to increase air flow through the foam and thus increase the evaporation rate of the working fluid.

An evaporatively cooled cold pack could also be made with the carbon foam. It would be somewhat similar to those currently available that are frozen prior to use, and may be fabricated using the carbon foam material. A carbon foam block would be encapsulated with a impermeable material. The working fluid would be poured in, wet the foam surface and evaporate, causing the foam temperature to drop. An opening through which the working fluid would be poured would also allow the evaporating fluid to vent to atmosphere.

Fig. 19 shows the carbon foam 118 of this invention in the form of a block 51 to be used as an automobile radiator generally 50. Hot engine cooling fluid is introduced into intake manifold 52 connected to pipes 54 which pass through the foam block 51 to the output manifold 56. As seen in Fig. 20 foam block 51 is supported in an automobile as indicated at 58 having the usual frame 53 and wheels 55. Hot fluid is conveyed by output conduit 57 from engine 59 to intake manifold 52. Cooled fluid returns to engine 59 by intake conduit 60 from output manifold 56. As the automobile 58 is moving down the road, air is forced through the foam block 51 and removes the heat to the environment. The efficiency of heat transfer from the radiator 50 to the ambient air is directly related to the surface area of the

block 51. Since a foam block 2 feet by 2 feet by 1 inches has approximately 19,000 m<sup>2</sup> of surface area while a typical radiator may approach 10 m<sup>2</sup>, the increased efficiency of the radiator will improve by roughly 3 orders of magnitude.

Fig. 21 shows the carbon foam 118 in the form of a spinning disk device generally 70. The disk device includes a foam disk portion 72 connected with a double walled conduit 74 providing a central hollow conduit member 76 and an outer conduit member 78. Air and an evaporative fluid are introduced into conduit 78 where it passes into the foam disk portion 72. The air and evaporative fluid are spun out of the disk portion 72 as it is rotated to the outside of the disk portion 72. This is shown by arrow 80 for the air and arrow 82 for the evaporative liquid. A fluid impermeable coating 79 provides a sealed surface on opposing sides of the disk portion 72. Hot fluid to be cooled is passed down central hollow conduit member 76 where it is cooled in disk portion 72. It flows out the bottom of conduit 76 as indicated at 84. The spinning disk portion 72 is supported by the bearings 86 and 88 in a suitable housing. Rotation of disk portion 72 is effected by motor 90 driving pulleys 92 and 94 by drive belt 96 with pulley 94 connected to conduit 74.

It will thus be seen that through the present invention there is provided:

- (i) A carbon foam having a very high thermal conductivity. Large temperature gradients are thus unlikely to develop, and the surface cooling due to evaporation will be quickly translated to bulk material cooling.
- (ii) The foam has an extended surface area resulting from its cellular structure. This allows for rapid evaporation of the working fluid.
- (iii) The foam has an open structure which allows the working fluid to permeate the material.

- (iv) The cell size and ligament properties may be varied, allowing the material to be tailored to the selected working fluid or anticipated cooling application.
- (v) A working fluid may be selected that is non-toxic and environmentally acceptable.
- (vi) Evaporative cooling systems such as those disclosed herein potentially offers low  
5 (zero) energy consumption and increased reliability with few or no moving parts.

Another object of the present invention is the production of a carbon foam heat sink product, i.e., encased high thermal conductivity porous carbon foam filled with a phase change material wherein tremendous amounts of thermal energy are stored and emitted very rapidly. The porous foam is filled with a phase change material (PCM) at a temperature close  
10 to the operating temperature of the device. As heat is added to the surface, from a heat source such as a computer chip, friction due to re-entry through the atmosphere, or radiation such as sunlight, it is transmitted rapidly and uniformly throughout the foam and then to the phase change material. As the material changes phase, it absorbs orders of magnitude more energy than non-PCM material due to transfer of the latent heat of fusion or vaporization.

15 Conversely, the filled foam can be utilized to emit energy rapidly when placed in contact with a cold object.

Non-limiting embodiments disclosed herein are a device to rapidly thaw frozen foods or freeze thawed foods, a design to prevent overheating of satellites or store thermal energy as they experience cyclic heating during orbit, and a design to cool leading edges during  
20 hypersonic flight or re-entry from space.

In order to illustrate the carbon foam heat sink product of this invention, the following examples are set forth. They are not intended to limit the invention in any way.

## EXAMPLE XVII: Device for Thawing Food

Acetic acid has a heat of melting of 45 J/g at a melting point of 11° C. The heat of melting of food, primarily ice, is roughly 79 J/g at 0°C. Therefore, take a block of foam and fill it with liquid acetic acid at room temp. The foam will be encased in a box made from an insulating polymer such as polyethylene on all sides except the top. The top of the foam/acetic acid block will be capped with a high thermal conductivity aluminum plate that snaps into place thus sealing the foam/acetic acid inside the polymer case (illustrated in figure 22). If the foam block is 10-in. x 15-in. x 0.5-in. thick, the mass of foam is 614 grams. The mass of acetic acid that fills the foam is roughly 921 grams. Therefore, when a piece of frozen meat is placed in contact with the top of the aluminum block, the foam will cool to the freezing point of the acetic acid (11°C). At this point, the heat given off from the acetic acid as it freezes (it also remains at 11°C) will be equivalent to 49 KJ. This heat is rapidly transferred to the frozen meat as it thaws (it also remains at 0°C). This amount of heat is sufficient to melt roughly 500 grams (1 lb.) of meat.

## EXAMPLE XVIII: Heat Sink To Prevent Overheating Of Satellites During Cyclic Orbits.

Produce a carbon-carbon composite with the foam in which the foam is a core material with carbon-carbon face sheets (figure 23). Fill the foam core with a suitable phase change material, such as a paraffin wax, that melts around the maximum operating temperature of the satellite components. One method to perform this would be to drill a hole in one surface of the carbon-carbon face sheets and vacuum fill the phase change material in the liquid state into the porous foam. Once filled, the sample can be cooled (the phase change material solidifies) and the hole can be plugged with an epoxy or screw-type cap. The epoxy and any other sealant must be able to withstand the operating temperature of the application. The foam-core composite will then be mounted on the side of the satellite that is exposed to



the sun during orbit. As the satellite orbits the earth and is exposed to the sun, the radiant energy from the sun will begin to heat the composite panel to the melting point of the phase change material. At this point, the panel will not increase in temperature as the phase change material melts. The amount of radiant energy the panel can absorb will be dependent on the thickness and outer dimensions of the panel. This can be easily calculated and designed through knowledge of the orbit times of the satellite such that the material never completely melts and, thus, never exceeds the melt temperature. Then, when the satellite leaves the view of the sun, it will begin to radiate heat to space and the phase change material will begin to freeze. The cycle will repeat itself once the satellite comes into view of the sun once again.

#### 10 EXAMPLE XIX: Heat Sink for Leading Edges

Currently, the shuttle orbiter experiences extreme heats during reentry. Specifically, the leading edges of the craft can reach 1800°C and the belly of the craft can reach temperatures as high as 1200°C. If a foam core composite panel is placed at the surface of the leading edges and at the surface of the belly (Figure 24), it would be able to absorb enough energy to dramatically reduce the maximum temperature of the hot areas. This also would permit a faster re-entry or (steeper glide slope) and maintain the current maximum temperatures. In this case the phase change material would most likely be an alloy, e.g. germanium-silicon, which melts around 800-900°C and does not vaporize until much higher than the maximum temperature of the craft.

20 For example, Germanium has a heat of formation (heat of melting) of 488 J/g. This would require 1.0 Kg of Germanium to reduce the temperature of 1 Kg of existing carbon/carbon heat-shield by 668°C. In other words, if the existing carbon-carbon were replaced pound-for-pound with germanium filled foam, the maximum temperature of the heat

shield would only be about 1131°C instead of about 1800°C during re-entry, depending on the duration of thermal loading.

This U.S. patent application incorporates by reference in their entireties the following:  
U.S. patent application no. 08/921,875 filed September 2, 1997; U.S. patent application no.  
5 08/923,877 filed September 2, 1997; U.S. patent application no. 09/093,406 filed June 8,  
1998; and U.S. patent application no. 09/458,640 filed December 9, 1999; Klett, J., "High  
Thermal Conductivity, Mesophase Pitch-Derived Graphitic Foam," 43<sup>rd</sup> Int'l SAMPE  
Symposium, May 31-June 4, 1998 (Aneheim, CA); J. Klett, C. Walls and T. Burchell, "High  
Thermal Conduc-tivity Mesophase Pitch-Derived Carbon Foams: Effect of Precursor on  
10 Structure and Properties," Carbon '99, 24<sup>th</sup> Biennial Conference on Carbon July 11-16, 1999;  
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Composites in Mfg., 15:4, pp.1-7 (1999); J. Klett, and T. Burchell, "High Thermal  
Conductivity, Mesophase Pitch Derived Carbon Foam," Science and Technology of Carbon,  
Extended Abstracts and Eurocarbon Programme, vol. II, Strasbourg, France, July 5-9, 1998;  
15 and pages published at the web site of Poco Graphite, Inc. of Decatur, Texas, at the Internet  
address poco.com/pocofoam/grafprod, as downloaded on January 21, 2000.

**Thermal Conductivity and Specific Thermal Conductivity:** The validity of the flash  
diffusivity method and whether the open porosity would permit penetration of the heat pulse  
into the sample had to be established. Deep penetration of the pulse in samples typically  
20 causes a change in the characteristic heat pulse on the back face of the sample. Thus, errors in  
the reported diffusivity can be as high as 20%. However, the rather large struts and small  
openings of the foam limits the depth of penetration to about one to two pore diameters (250 -  
500 micrometers), or less than 2% penetration. Therefore, it was believed that this technique

would yield a fairly accurate value for the thermal conductivity. This was confirmed by testing samples with both the flash diffusivity method and the thermal gradient method. The measured conductivities varied by less than 5%, verifying the flash method as a viable method to measure these foams. If the pore structure changes significantly, the flash method will likely yield inaccurate results.

In another embodiment of the invention, two different precursors were used to produce foam with the process of the invention. These precursors were a Conoco Mesophase Pitch and a Mitsubishi ARA24 Mesophase Pitch (herein referred to as Conoco and ARA24). The results are shown in Tables II and III. They were processed with varying operating pressures under nitrogen atmosphere, a heating rate during the foaming step of 3.5°C/min, coked at 630°C for 1 hour, and cooled at the natural cooling rate of the furnace. The samples were carbonized in a separate furnace under nitrogen at a heating rate of 0.2°C/min up to 1000°C and then some samples were graphitized at 2800°C in yet another furnace at two different heating rates (10°C/min and 4°C/min, Table III).

The thermal conductivity (a term which is herein used synonymously with "bulk thermal conductivity") of the foam was very high as shown in Table II and Figures 25 and 26. The thermal conductivity of the graphitized ARA24 foam, graphitized at 4°C/min, was in the range of approximately 146 to 187 W/m·K, as shown in Table III. This is remarkable for a material with such a low density of approximately 0.56 g/cm<sup>3</sup>. This calculates as a specific thermal conductivity (thermal conductivity divided by the density) in the range of approximately 256 to 334 W·cm<sup>3</sup>/m<sup>3</sup>·K·gm. As stated earlier, for a foam with a bulk thermal conductivity of approximately 58 W/m·K, the ligament thermal conductivity is approximately 700 W/m·K. However, with the data shown in Tables II and III, when the thermal conductivity of the foam is about 147 W/m·K, the ligament thermal conductivity is

approximately 1800 W/m·K and for a foam thermal conductivity approximately 187 W/m·K, the ligament thermal conductivity is approximately 2200 W/m·K.

It is an unusual property of the invention that the thermal conductivity of this graphitic carbon foam is substantially isotropic, and is preferably completely isotropic. The foam exhibits substantially isotropic thermal conductivity comparable to the isotropic thermal conductivity of some metallic thermal management materials (Table IV). The foam exhibits a thermal conductivity, 146 to 187 W/m·K for the ARA24 foam in Table III, that is comparable to the in-plane thermal conductivity of other carbon-based thermal management materials, such as the carbon-carbon composites containing carbon fiber that are listed in Table IV, which are 109 W/m·K and 250 W/m·K. The foam has a significantly higher thermal conductivity in the out-of-plane direction than these carbon-carbon composites, which are 1 W/m·K and 20 W/m·K. Carbon-based thermal management materials typically exhibit substantial differences between in-plane and out-of-plane thermal conductivity, as shown in Table IV. Although several of the other thermal management materials have higher in-plane thermal conductivities, their densities are much greater than the foam, i.e., the specific thermal conductivity of the foam is significantly greater than all the available thermal management materials. In fact, the specific thermal conductivity is more than seven times greater than copper ( $45 \text{ W}\cdot\text{cm}^3/\text{m}\cdot\text{K}\cdot\text{gm}$ ), the preferred material for heat sinks in the prior art. It is clear that, for thermal management, where weight is a concern or where un-steady state conditions occur often, the graphitic foam is superior to most other available materials. The advantage of isotropic thermal and mechanical properties should allow for novel designs that are more flexible and more efficient.

Table II. Properties of Carbon Foam Samples of the Invention

Table II. Properties of Carbon Foam Samples of the Invention

Sample ID	Pitch Precursor	Max Heat Treatment Temperature [°C]	Foaming Pressure [Psi]	Total Pore Area [m <sup>2</sup> /g]	Mean Pore Diameter* [microns]	Density [g/cm <sup>3</sup> ]	Specific Surface Area [m <sup>2</sup> /m <sup>3</sup> ]	Thermal Conductivity** [W/m·K]	Specific Thermal Conductivity [W·cm <sup>3</sup> /m·K·gm]
G	AR	1000	400	81.979		0.22	13,635,380	0.6	2.7
A	AR	1000	600	47.89	125	0.37	17,719,300	1.2	3.2
M	AR	1000	800	70.31	168	0.44	30,926,400	1.3	3.0
P	AR	1000	1000	0.026	90.7	0.54	19,440	1.7	3.1
F	Conoco	1000	400	0.956	59.44	0.33	315,480	0.9	2.7
E	Conoco	1000	600	0.166	46.93	0.4	66,400	1	2.5
D	Conoco	1000	800	20.317	28.8	0.49	9,955,330	1.3	2.7
B	Conoco	1000	1000	20.565	24	0.56	11,516,400	1.2	2.1
N	AR	2800	400	0.025	340	0.25	6,250	50	200.3
K	AR	2800	600	112.4	165	0.39	43,826,000	72	184.3
L	AR	2800	800	60.81	100.2	0.48	29,189,300	105	213.3
O	AR	2800	1000	0.045	100.85	0.57	25,550	149	261.4
Q	AR	2800	1000			0.57			
I	Conoco	2800	400	0.087	59.19	0.35	30,450	40.8	116.5
J	Conoco	2800	600	0.162	48.45	0.4	64,300	85.1	212.3
H	Conoco	2800	800	0.15	41.23	0.49	73,500	104.2	212.7
C	Conoco	2800	1000	27.06	31.3	0.59	15,965,400	134.1	227.3

\*note: The mean pore diameter, since it was calculated from results by mercury porosimetry, is not representative of average cell size.

\*\*note: Thermal conductivity was calculated from the measured thermal diffusivity using a xenon pulse flash diffusivity technique.

Table III. Thermal Conductivity and Specific Thermal Conductivity vs. Density for Mesophase Derived Graphite Foams made from Different Precursors in the Invention.

Conoco				ARA24			
Density [g/cm <sup>3</sup> ]	Thermal Conductivity [W/m-K]	*Specific Thermal Conductivity [W-cm <sup>3</sup> /m-K-gm]	Graphitization Rate [°C/min]	Density [g/cm <sup>3</sup> ]	Thermal Conductivity [W/m-K]	*Specific Thermal Conductivity [W-cm <sup>3</sup> /m-K-gm]	Graphitization Rate [°C/min]
0.59	134.1	227	10	0.56	187	334	4
0.56	92.1	164	10	0.59	183	310	4
0.56	80.1	143	10	0.62	180	290	4
0.54	102	189	10	0.56	177	316	4
0.53	99	187	10	0.58	170	293	4
0.49	104.2	213	10	0.61	169	277	4
0.43	85.2	198	10	0.56	166	296	4
0.4	84.1	210	10	0.56	165	295	4
0.4	85.1	213	10	0.6	161	268	4
0.32	55.1	172	10	0.61	160	262	4
0.36	40.9	114	10	0.59	157	266	4
0.35	40.8	117	10	0.62	152	245	4
				0.59	151.2	256	4
				0.6	150	250	4
				0.57	148.9	261	4
				0.57	146	256	4
ARA24							
Density [g/cm <sup>3</sup> ]	Thermal Conductivity [W/m-K]	*Specific Thermal Conductivity [W-cm <sup>3</sup> /m-K-gm]	Graphitization Rate [°C/min]	Density [g/cm <sup>3</sup> ]	Thermal Conductivity [W/m-K]	*Specific Thermal Conductivity [W-cm <sup>3</sup> /m-K-gm]	Graphitization Rate [°C/min]
0.6	136	227	10	0.52	93	179	10
0.6	131.6	219	10	0.38	92.2	243	10
0.51	127	249	10	0.45	86.8	193	10
0.53	127	240	10	0.55	85.3	155	10
0.52	121	233	10	0.4	75	188	10
0.47	119.6	254	10	0.39	74.5	191	10
0.53	118	223	10	0.39	68.2	175	10
0.57	112	196	10	0.29	67.1	231	10
0.48	105.3	219	10	0.28	62	221	10
0.48	104.5	218	10	0.31	55.2	178	10
0.57	98	172	10	0.3	52.6	175	10
0.51	98	192	10	0.25	50.5	202	10
0.49	94	192	10	0.27	48.3	179	10

\* Specific Thermal Conductivity = Thermal Conductivity divided by Density.

Table IV.

Material	Thermal Conductivity			Specific Thermal Conductivity	
	Density [gm/cm <sup>3</sup> ]	In-plane [W/m·K]	Out-of-plane [W/m·K]	In-plane [W·cm <sup>3</sup> /m <sup>3</sup> ·K·gm]	Out-of-plane [W·cm <sup>3</sup> /m <sup>3</sup> ·K·gm]
Typical 2-D Carbon-Carbon	1.88	250	20	132	10.6
EW-C-300/Cyanate Ester	1.172	109	1	63	0.6
Copper	8.9	400	400	45	45
Aluminum 6061	2.8	180	180	64	64
Aluminum Honeycomb	0.19	--	~10	--	52
Aluminum Foam	0.5	12	12	24	24

Based on the data in Tables II, III, and IV and in Figures 25 and 26, it can be seen that the carbon foams of the invention, when graphitized, have surprisingly high thermal conductivity and specific thermal conductivity. The graphitic carbon foams typically have a thermal conductivity of at least 40 W/m·°K and/or a specific thermal conductivity at least equal to copper, i.e., at least 45 W·cm<sup>3</sup>/m<sup>3</sup>·°K·gm, and usually on the order of at least 75 W·cm<sup>3</sup>/m<sup>3</sup>·°K·gm. More typical graphitic carbon foams of the invention have a thermal conductivity of at least 75 W/m·°K and/or a specific thermal conductivity of at least 100 W·cm<sup>3</sup>/m<sup>3</sup>·°K·gm. In the preferred embodiment, the carbon foams of the invention have a thermal conductivity of at least 100 W/m·°K and/or a specific thermal conductivity of at least 150 W·cm<sup>3</sup>/m<sup>3</sup>·°K·gm. More preferred embodiments contemplate carbon foams having a thermal conductivity of at least 125 W/m·°K and/or a specific thermal conductivity of at least 175 W·cm<sup>3</sup>/m<sup>3</sup>·°K·gm. Yet more preferred embodiments contemplate carbon foams having a thermal conductivity of at least 150 W/m·°K and/or a specific thermal conductivity of at least

200 W·cm<sup>3</sup>/m·°K·gm. Still more highly preferred embodiments contemplate carbon foams having a thermal conductivity of at least 175 W/m·°K and/or a specific thermal conductivity of at least 250 W·cm<sup>3</sup>/m·°K·gm, with the data in Tables II-IV and Figures 25 and 26 showing that specific thermal conductivities on the order of at least 275 W·cm<sup>3</sup>/m·°K·gm, and even at  
 5 least 300 W·cm<sup>3</sup>/m·°K·gm, and indeed even at least 325 W·cm<sup>3</sup>/m·°K·gm, are at-tainable. Pore diameters indicated in Table II were measured by the mercury porosimetry method.

**Specific Surface Area:** Another property that affects the overall thermal performance of the carbon foam is the specific surface area (SSA), calculated by:

$$SSA [m^2/m^3] = Total Pore Area [m^2/g] \times Density [g/cm^3] \times 1,000,000 [cm^3/m^3]$$

10 Smaller specific surface areas indicate a lower foam porosity which reduces the effect of the natural convective heat transfer mode (laminar flow) and allows the more efficient conductive heat transfer mode to dominate thermal performance. Larger SSA's enhance evaporative cooling via increased surface area to volume ratio and increasing the contact area between the evaporative fluid and the foam material. SSA is also an indicator of the foam's response to  
 15 forced convective heat transfer (turbulent flow) via fluid passing through the media by increasing the surface area used for heat transfer.

As shown in Table II, the SSA value for the graphitized carbon foams of the invention (heat treated to 2800° C.) was at least about 6,000 m<sup>2</sup>/m<sup>3</sup>, typically above 25,000 m<sup>2</sup>/m<sup>3</sup>, and even more typically above 65,000 m<sup>2</sup>/m<sup>3</sup>. SSA values of at least 100,000 m<sup>2</sup>/m<sup>3</sup> or at least  
 20 500,000m<sup>2</sup>/m<sup>3</sup> are contemplated in the invention. Indeed, several samples shown in Table II were above 1,000,000 m<sup>2</sup>/m<sup>3</sup>, and in the preferred embodiment, the carbon foam has an SSA



of at least about 2,000,000 m<sup>2</sup>/m<sup>3</sup>, more preferably at least about 5,000,000 m<sup>2</sup>/m<sup>3</sup>, more preferably still at least about 10,000,000 m<sup>2</sup>/m<sup>3</sup>, and most preferably, at least about 15,000,000 m<sup>2</sup>/m<sup>3</sup>, with SSA values of at least about 25,000,000 m<sup>2</sup>/m<sup>3</sup> or at least about 35,000,000 also being contemplated. The upper possible limit on the SSA value is currently  
5 unknown, and while the data in Table II show the highest value achieved with the few samples therein tested as 43,836,000 m<sup>2</sup>/m<sup>3</sup>, values both higher or lower than this value are contemplated as within the invention.

**Evaporative Cooling:** Examples VII, VIII, and X, along with Figure 14 show that the still air experiments in a vacuum furnace produced the following cooling rates:

- 10           I.           Using acetone as the fluid, the carbon foam temperature reached -53.4°C (C) in no more than about 4 minutes;
- II.           Using ethanol as the fluid, the carbon foam temperature reached -28.6°C in no more than about 6 minutes;
- III          Using water as the fluid, the carbon foam temperature reached -5.5°C  
15                       in no more than about 3 minutes.

Examples XI, XII, and XIII, along with Figure 15, show that the natural convection experiments conducted under ambient room temperature conditions produced the following cooling rates:

- I.           Using acetone as the fluid, the carbon foam temperature reached 0.5°C in no  
20                       more than 38 minutes;
- II.           Using ethanol as the fluid, the carbon foam temperature reached 14.3°C in no more than 38 minutes; and

III. Using water as the fluid, the carbon foam temperature reached 18.9°C in no more than 38 minutes.

As indicated above, the foregoing information was derived from the data presented with respect to the experiments previously described showing decreases in temperature with time using acetone, ethanol, and water, respectively. But these are by no means the only conclusions that can be drawn from the tabular data set forth for these experiments. For example, in Example VII it is shown, under the conditions of the experiment, that, when acetone was the fluid, the carbon foam reached a temperature of -46.7°C in no more than 2 minutes at a reduced pressure (vacuum) of 29 torr. Similarly, in Experiment VIII it is shown, under the conditions of the experiment, that, when ethanol was the fluid, the carbon foam reached a temperature of -21.7°C in no more than 3 minutes at a reduced pressure (vacuum) of 1.2 torr. And again similarly, in Experiment XI, with acetone as a fluid, and under the natural convection conditions of the experiment, the carbon foam reached a temperature of 5.7°C in no more than 10 minutes. These and similar conclusions may be drawn from the tabular data in Examples VII to XVI, which data illustrate the combined effects the high thermal conductivity and SSA properties of the carbon foam of the invention have upon the cooling rate achievable with acetone, ethanol, and water, respectively.

**X-ray Analysis:** Lattice parameters were determined from the indexed diffraction peak positions (Table V). The X-ray method for crystallite size determination is well known to those skilled in the art. The 002 and 100 diffraction peak breadths were analyzed using the Scherrer equation to determine the crystallite dimensions in the a- and c- directions.

$$t = \frac{0.9\lambda}{B \cos(2\theta)}$$

where  $t$  is the crystallite size,  $\lambda$  is the X-ray wavelength,  $B$  is the breadth of the diffraction peak [full width half maximum (FWHM) minus the instrumental breadth], and  $2\theta$  is the diffraction angle. As shown in Table V, the 002 peak (which is characteristic of interlayer

5 interlayer spacing calculated with the Scherrer method is in the range of approximately 0.3354 nm to 0.3364 nm. The crystallite size in the c-direction was calculated from these data to be at least approximately 82.4 nm, and the 100 peak (or 1010 in hexagonal nomenclature) was used to calculate the crystallite size in the a-direction of at least approximately 21.5 nm. These crystallite sizes are larger than typical high thermal conductivity carbon fibers and

10 therefore, the foam ligaments should perform similarly to high order pyrolytic carbon and high thermal conductivity carbon fibers such as K1100 and vapor grown carbon fibers (VGCF).

Table V. X-ray Diffraction Data for Carbon Foam Samples

*Sample ID	Pitch Precursor	Max Heat Treatment Temperature (°C)	Foaming Pressure (Psi)	$d_{002}$ spacing (nm)	$L_a$ (nm)	$L_c$ (nm)	Peak Angles (2 $\theta$ )			FWHM			Narrowness Relative Peak Split Factor (RPSF)
							002	101	100	002	101	100	
original	AR	1000	1000	0.3362	20.3	44.2	26.4853	42.3185	44.2751	0.2940	0.5540	1.2870	0.470
N	AR	2800	400	0.3364	11.8	48.2	26.4769	42.1512	44.1507	0.2292	0.7644	0.8856	0.413
K	AR	2800	600	0.3362	17.8	46.6	26.4839	42.0911	44.2000	0.2348	0.5374	0.7207	0.298
L	AR	2800	800	0.3360	21.5	79.3	26.5006	42.1416	44.2059	0.1628	0.4542	0.7807	0.299
O	AR	2800	1000	0.3356	21.4	56.7	26.5540	42.2270	44.2815	0.1590	0.5220	0.7438	0.308
Q	AR	2800	1000	0.3354	18.4	82.4	26.5393	42.3065	44.2815	0.2040	0.4568	0.7438	0.304

\* Samples N-Q of Table V are the same as samples N-Q of Table II, respectively.

15 The "doublet" at the 100 and 101 peaks is characterized by a relative peak split factor (RPSF) parameter, or narrowness, calculated using the peak angles and the full width half maximums (FWHM). The equation is:

$$RPSF = \frac{\left( \frac{FWHM_{101}}{2} + \frac{FWHM_{100}}{2} \right)}{2\theta_{100} - 2\theta_{101}}$$

A smaller RPSF indicates closer peaks at 100 and 101 and favorable lattice conditions for thermal conductivity and structural integrity. The data reported in Table V shows values for RPSF no greater than 0.470, with the lowest reported value for a carbon foam heat treated at 2800°C being 0.298 and the highest being 0.413.

5     **Microstructural Characterization:** Figures 27 and 28 are scanning electron micrographs of the pore structure of a foam sample of the invention. The foam exhibits a structure having open interconnects between cells (or pores) 200, with such cells (or pores) being of similar geometric shape, typically ellipsoidal, and sometimes spherical or essentially spherical. (It is noted that a sphere is a specific form of an ellipsoid.) It is evident from the images that the  
10     graphitic structure is oriented parallel to the cell walls and highly aligned along the axis of ligaments 210. This highly aligned structure is significantly different from typical vitreous carbon foams: vitreous carbon foams are void of graphitic structure, have large openings and linear ligaments, and are mostly pentagonal dodecahedral in shape.

Moreover, it can be seen that in the junctions 220 between ligaments, the graphitic  
15     structure is less aligned and possesses more folded texture. It is postulated that this arises from the lack of stresses at this location during forming. Therefore, the well-ordered structure in these regions is primarily an artifact of the structure in precursor mesophase prior to heat treatment.

We claim:

1           1.     A carbon foam which, when initially at a room temperature of approximately  
2     21° C and saturated with acetone at said room temperature, cools to about -50° C in no more  
3     than about 4 minutes when placed under a reduced pressure (vacuum) approximately  
4     equivalent to 1.2 torr.

1           2.     A carbon foam which, when initially at a room temperature of approximately  
2     21° C and saturated with ethanol at said room temperature, cools to about -28° C in no more  
3     than about 6 minutes when placed under a reduced pressure (vacuum) approximately  
4     equivalent to 1.0 torr.

1           3.     A carbon foam which, when initially at a room temperature of approximately  
2     21° C and saturated with water at said room temperature, cools to below 0° C in no more than  
3     about 3 minutes when placed under a reduced pressure (vacuum) approximately equivalent to  
4     29 torr.

1           4.     A carbon foam which, when initially at a room temperature of approximately  
2     21° C and saturated with acetone at said room temperature, cools to about 1.0° C in no more  
3     than about 34 minutes when exposed to ambient conditions under natural evaporation.

1           5.     A carbon foam which, when initially at a room temperature of approximately  
2     21° C and saturated with ethanol at said room temperature, cools to about 15° C in no more  
3     than about 23 minutes when exposed to ambient conditions under natural evaporation.

1           6.     A carbon foam which, when initially at a room temperature of approximately  
2     21° C and saturated with water at said room temperature, cools to about 19° C in no more  
3     than about 33 minutes when exposed to ambient conditions under natural evaporation.

1           7.     A carbon foam having a thermal conductivity of at least 40 W/m·°K.

1           8.     A carbon foam as defined in claim 7 having a thermal conductivity of at least  
2     50 W/m·°K.

1           9.     A carbon foam as defined in claim 7 having a thermal conductivity of at least  
2     75 W/m·°K.

1           10.    A carbon foam as defined claim 7 having a thermal conductivity of at least 100  
2     W/m·°K.

1           11.    A carbon foam as defined in claim 7 having a thermal conductivity of at least  
2     125 W/m·°K.

1           12.    A carbon foam as defined in claim 7 having a thermal conductivity of at least  
2     150 W/m·°K.

1           13.    A carbon foam as defined in claim 7 having a thermal conductivity of at least  
2     175 W/m·°K.

1           14.    A carbon foam as defined in claim 8, 10, 11, or 13 having a maximum thermal  
2   conductivity of about  $187 \text{ W/m}^\circ\text{K}$ .

1           15.    A carbon foam as defined in claim 7 having a maximum thermal conductivity  
2   of about  $187 \text{ W/m}^\circ\text{K}$ .

1           16.    A carbon foam having a specific thermal conductivity of at least 75  
2    $\text{W}\cdot\text{cm}^3/\text{m}^\circ\text{K}\cdot\text{gm}$ .

1           17.    A carbon foam as defined in claim 7, 9, 13, or 15 having a specific thermal  
2   conductivity of at least  $100 \text{ W}\cdot\text{cm}^3/\text{m}^\circ\text{K}\cdot\text{gm}$ .

1           18.    A carbon foam as defined in claim 16 having a specific thermal conductivity  
2   of at least  $125 \text{ W}\cdot\text{cm}^3/\text{m}^\circ\text{K}\cdot\text{gm}$ .

1           19.    A carbon foam as defined in claim 7, 11, 15, or 16 having a specific thermal  
2   conductivity of at least  $150 \text{ W}\cdot\text{cm}^3/\text{m}^\circ\text{K}\cdot\text{gm}$ .

1           20.    A carbon foam as defined in claim 16 having a specific thermal conductivity  
2   of at least  $175 \text{ W}\cdot\text{cm}^3/\text{m}^\circ\text{K}\cdot\text{gm}$ .

1           21.    A carbon foam as defined in claim 7, 9, 13, 15, or 16 having a specific thermal  
2   conductivity of at least  $200 \text{ W}\cdot\text{cm}^3/\text{m}^\circ\text{K}\cdot\text{gm}$ .

1           22.    A carbon foam as defined in claim 16 having a specific thermal conductivity  
2   of at least  $225 \text{ W}\cdot\text{cm}^3/\text{m}\cdot^\circ\text{K}\cdot\text{gm}$ .

1           23.    A carbon foam as defined in claim 7, 11, 13, 15, or 16 having a specific  
2   thermal conductivity of at least  $250 \text{ W}\cdot\text{cm}^3/\text{m}\cdot^\circ\text{K}\cdot\text{gm}$ .

1           24.    A carbon foam as defined in claim 16 having a specific thermal conductivity  
2   of at least  $275 \text{ W}\cdot\text{cm}^3/\text{m}\cdot^\circ\text{K}\cdot\text{gm}$ .

1           25.    A carbon foam as defined in claim 7, 9, 13, 15 or 16 having a specific thermal  
2   conductivity of at least  $300 \text{ W}\cdot\text{cm}^3/\text{m}\cdot^\circ\text{K}\cdot\text{gm}$ .

1           26.    A carbon foam as defined in claim 16 having a specific thermal conductivity  
2   of at least  $325 \text{ W}\cdot\text{cm}^3/\text{m}\cdot^\circ\text{K}\cdot\text{gm}$ .

1           27.    A carbon foam as defined in claim 7, 11, 13, or 15 having a specific thermal  
2   conductivity in the range from about 114 to about  $334 \text{ W}\cdot\text{cm}^3/\text{m}\cdot^\circ\text{K}\cdot\text{gm}$ .

1           28.    A carbon foam as defined in claim 16 having a maximum specific thermal  
2   conductivity of about  $334 \text{ W}\cdot\text{cm}^3/\text{m}\cdot^\circ\text{K}\cdot\text{gm}$ .

1           29.    The carbon foam of claim 7 having a specific surface area of at least  $6000 \text{ m}^2/\text{m}^3$ .



1           30. The carbon foam of claim 15, 16, 28, or 29 having a specific surface area of at  
2           least 25,000 m<sup>2</sup>/m<sup>3</sup>.

1           31. The carbon foam of claim 29 having a specific surface area of at least 65,000  
2           m<sup>2</sup>/m<sup>3</sup>.

1           32. The carbon foam of claim 17 having a specific surface area of at least 65,000  
2           m<sup>2</sup>/m<sup>3</sup>.

1           33. The carbon foam of claim 13, 16, 28, or 29 having a specific surface area of at  
2           least 100,000 m<sup>2</sup>/m<sup>3</sup>.

1           34. The carbon foam of claim 27 having a specific surface area of at least 500,000  
2           m<sup>2</sup>/m<sup>3</sup>.

1           35. The carbon foam of claim 29 having a specific surface area of at least 500,000  
2           m<sup>2</sup>/m<sup>3</sup>.

1           36. The carbon foam of claim 29 having a specific surface area of at least 1,000,000  
2           m<sup>2</sup>/m<sup>3</sup>.

1           37. The carbon foam of claim 19 having a specific surface area of at least 1,000,000  
2           m<sup>2</sup>/m<sup>3</sup>.

1           38. The carbon foam of claim 29 having a specific surface area of at least 5,000,000  
2           m<sup>2</sup>/m<sup>3</sup>.

1           39. The carbon foam of claim 9, 15, 18, 28 or 29 having a specific surface area of at  
2           least 10,000,000 m<sup>2</sup>/m<sup>3</sup>.

1           40. The carbon foam of claim 29 having a specific surface area of at least 15,000,000  
2           m<sup>2</sup>/m<sup>3</sup>.

1           41. The carbon foam of claim 21 having a specific surface area of at least 20,000,000  
2           m<sup>2</sup>/m<sup>3</sup>.

1           42. The carbon foam of claim 9, 15, 20, or 29 having a specific surface area of at  
2           least 25,000,000 m<sup>2</sup>/m<sup>3</sup>.

1           43. The carbon foam of claim 29 having a specific surface area of at least 30,000,000  
2           m<sup>2</sup>/m<sup>3</sup>.

1           44. The carbon foam of claim 15, 28, or 29 having a specific surface area of at least  
2           35,000,000 m<sup>2</sup>/m<sup>3</sup>.

1           45. The carbon foam of claim 29, 36, or 38 having a specific surface area no greater  
2           than about 44,000,000 m<sup>2</sup>/m<sup>3</sup>.

1           46. The carbon foam of claim 34 having a specific surface area no greater than about  
2           44,000,000 m<sup>2</sup>/m<sup>3</sup>.

1           47. A carbon foam having an X-ray diffraction pattern characterized by doublet 100  
2           and 101 peaks characterized by a relative peak split factor no greater than about 0.470.

1           48. A carbon foam as defined in claim 33 having an X-ray diffraction pattern  
2 characterized by doublet 100 and 101 peaks characterized by a relative peak split factor no  
3 greater than about 0.470.

1           49. A carbon foam as defined in claim 39 having an X-ray diffraction pattern  
2 characterized by doublet 100 and 101 peaks characterized by a relative peak split factor no  
3 greater than about 0.470.

1           50. A carbon foam as defined in claim 21 having an X-ray diffraction pattern  
2 characterized by doublet 100 and 101 peaks characterized by a relative peak split factor no  
3 greater than about 0.470.

1           51. A carbon foam as defined in claim 8, 10, 12, 16, 20, 24, 29, 31, 35, or 40 having  
2 an X-ray diffraction pattern characterized by doublet 100 and 101 peaks characterized by a  
3 relative peak split factor no greater than about 0.470.

1           52. A carbon foam as defined in claim 46 having an X-ray diffraction pattern  
2 characterized by doublet 100 and 101 peaks characterized by a relative peak split factor  
3 between about 0.298 and about 0.413.

1           53. A carbon foam as defined in claim 27 having an X-ray diffraction pattern  
2 characterized by doublet 100 and 101 peaks characterized by a relative peak split factor no  
3 greater than about 0.413.

1           54.     A carbon foam as defined in claim 9, 11, 13, 18, 22, 26, 28, 36, or 38 having  
2     an X-ray diffraction pattern characterized by doublet 100 and 101 peaks characterized by a  
3     relative peak split factor between about 0.298 and about 0.470.

1           55.     A carbon foam having an X-ray diffraction pattern characterized by a full width  
2     half maximum for the 002 peak angle of between about 0.159 and about 0.294 degrees.

1           56.     A carbon foam as defined in claim 52 wherein said X-ray diffraction pattern is  
2     further characterized by a full width half maximum for the 002 peak angle of between about  
3     0.159 and about 0.2292 degrees.

1           57.     A carbon foam as defined in claim 53 wherein said X-ray diffraction pattern is  
2     further characterized by a full width half maximum for the 002 peak angle of between about  
3     0.159 and about 0.294 degrees.

1           58.     A carbon foam as defined in claim 8, 13, 20, 26, 28, 29, or 36 characterized by  
2     an X-ray diffraction pattern having a full width half maximum for the 002 peak angle of  
3     between about 0.159 and about 0.294 degrees.

1           59.     A carbon foam as defined in claim 58 in which the X-ray diffraction pattern is  
2     further characterized by

3           (a) doublet 100 and 101 peaks further characterized by a relative peak split factor  
4     between about 0.298 and about 0.470, and

5           (b) a d002 spacing between about 0.3354 and about 0.3364.

1           60.     A carbon foam as defined in claim 59 having a mean pore diameter between  
2     about 30 and about 60 microns, and a density between about 0.2 and about 0.65 gm/cm<sup>3</sup>, and  
3     a relative peak split factor no greater than about 0.413.

1           61.     A carbon foam as defined in claim 1, 5, 7, 16, or 29 having a mean pore  
2     diameter no greater than 60 microns and a density between about 0.2 and about 0.65 gm/cm<sup>3</sup>.

1           62.     A heat exchange device comprising as a heat transfer surface the carbon foam  
2     of claim 7.

1           63.     An evaporatively cooled heat sink comprising the carbon foam of claim 7.

1           64.     The carbon foam of claim 7 containing micropores containing a phase change  
2     material.

1           65.     A carbon foam as defined in claim 7 having a density from about 0.4 to about  
2     0.65 gm/cm<sup>3</sup>.

1           66.     A carbon foam as defined in claim 7 having a density from about 0.55 to about  
2     0.60 gm/cm<sup>3</sup>.

1           67.     A carbon foam as defined in claim 7 having a density from about 0.50 to about  
2     0.65 gm/cm<sup>3</sup>.

1           68.    A carbon foam as defined in claim 7 having a mean pore diameter no greater  
2   than about 340 microns.

1           69.    A carbon foam as defined in claim 7 having a mean pore diameter no greater  
2   than about 60 microns.

1           70.    A carbon foam as defined in claim 7 characterized by an X-ray diffraction  
2   pattern exhibiting relatively sharp doublet peaks at  $2\theta$  angles between 40 and 50 degrees.

1           71.    A carbon foam as defined in claim 7 characterized by an X-ray diffraction  
2   pattern having an average d002 spacing of about 0.336.

1           72.    A carbon foam as defined in claim 7 characterized by an X-ray diffraction  
2   pattern having an average d002 spacing of 0.336.

1           73.    A carbon foam as defined in claim 7 having an open cell pore structure  
2   substantially comprised of ellipsoidal pores.

1           74.    A carbon foam as defined in claim 7 having an open cell pore structure  
2   consisting essentially of substantially ellipsoidal pores.

1           75.    A carbon foam as defined in claim 7 having an open cell pore structure  
2   comprised of pores whose planar cross-sectional images are substantially circular or elliptical.

1           76.    A carbon foam as defined in claim 7 having an open cell pore structure  
2   consisting essentially of pores whose planar cross-sectional images are substantially circular  
3   or elliptical.

1           77.    A carbon foam as defined in claim 7 of substantially open cell structure  
2   comprising cell walls wherein carbon derived from a mesophase pitch is substantially aligned  
3   along the cell wall axes, said carbon foam having been non-oxidatively stabilized.

1           78.    A carbon foam as defined in claim 77 wherein said carbon substantially  
2   aligned along the cell wall axes is substantially in the form of graphite.

1           79.    A carbon foam as defined in claim 7 characterized by being essentially  
2   completely graphitic and being further characterized by an open cell structure substantially  
3   comprised of ellipsoidal pores and by graphite substantially aligned along the axes of the cell  
4   walls.

1           80.    A carbon foam as defined in claim 7 having a pore structure consisting  
2   essentially of pores having diameters within a 100 micron range.

1           81.    A carbon foam as defined in claim 7 having an X-ray diffraction pattern  
2   substantially as depicted in Figure 11.

1           82.    A thermally conductive carbon foam having a substantially isotropic thermal  
2   conductivity.

1           83.     A carbon foam as defined in claim 82 wherein said thermal conductivity is at  
2     least about 40 W/m·°K.

1           84.     A carbon foam as defined in claim 82 wherein the specific thermal  
2     conductivity of said foam is greater than about 75 W·cm<sup>3</sup>/m·K·gm.

1           85.     A carbon foam as defined in claim 82 wherein said foam has a specific surface  
2     area of about 6000 m<sup>2</sup>/m<sup>3</sup>.

1           86.     A carbon foam as defined in claim 82 having at least about 90% of its pores  
2     congregated within a 100 micron pore diameter.

1           87.     A carbon foam as defined in claim 82 wherein said foam has a mean pore  
2     diameter no greater than about 340 microns.

1           88.     A carbon foam as defined in claim 82 wherein said foam has substantially  
2     ellipsoidal pores.

1           89.     A process for producing a carbon foam comprising  
2             (1) liquefying a pitch and  
3             (2) heating the liquefied pitch under non-oxidizing, superatmospheric conditions to  
4     produce a carbon foam.



1           90.    The process of claim 89 wherein the pitch is liquified in step (1) by heating  
2   under non-oxidizing conditions including a pressure lower than the superatmospheric  
3   pressure producing the carbon foam in step (2).

1           91.    The process of claim 89 wherein the pressure of said superatmospheric  
2   conditions producing said carbon foam in step (2) is less than about 69 atm (1000 psi).

1           92.    The process of claim 89 wherein the heating in step (2) is carried out under a  
2   static superatmospheric pressure.

1           93.    The process of claim 89 wherein said heating in step (2) is carried out at a  
2   temperature no greater than about 480 °C.

1           94.    The process of claim 89 wherein said heating of the liquified pitch under non-  
2   oxidizing, superatmospheric conditions in step (2) comprises heating at a temperature  
3   between about 420 °C. and about 480 °C.

1           95.    The process of claim 94 wherein the heating in step (2) is carried out such that  
2   at least some foaming occurs at said temperature between about 420 °C. and about 480 °C.

1           96.    The process of claim 89 wherein the heating in step (2) is carried out with no  
2   release of pressure until the formation of the carbon foam is substantially complete.

1           97.    The process of claim 89 further comprising  
2                   (3) coking said carbon foam by heating said carbon foam under non-oxidizing,  
3   superatmospheric conditions.

1           98.    The process of claim 97 wherein said heating in step (3) is carried out at a  
2   temperature above 500°C.

1           99.    The process of claim 97 wherein the temperature in step (3) is held constant  
2   for at least 15 minutes.

1           100.   The process of claim 97 wherein the pressure is held substantially constant  
2   during each of steps (2) and (3).

1           101.   The process of claim 97 wherein the pressure in steps (2) and (3) is identical.

1           102.   The process of 97 wherein the temperature during coking in step (3) is no  
2   more than 1000 °C.

1           103.   The process of claim 97 further comprising, after coking in step (3):  
2                   (4) carbonizing the carbon foam by heating in a non-oxidizing environment at  
3   a temperature higher than that for coking in step (3).

1           104.   The process of claim 103 wherein the coked foam is cooled prior to being  
2   carbonized in step (4).

1           105.   The process of claim 104 further comprising, after carbonizing in step (4):  
2                   (5) further heating at a yet higher temperature to yield a graphitic carbon foam.

1           106.   The process of claim 97 further comprising, after coking in step (3), heating  
2   under conditions sufficient to produce a graphitic carbon foam without oxidatively stabilizing  
3   the carbon foam.

1           107.   A pitch-derived carbon foam with an open cell pore structure substantially  
2   comprised of pores of similar geometric shape.

1           108.   A carbon foam as defined in claim 107 having struts interconnected  
2   throughout the carbon foam.

1           109.   A carbon foam as defined in claim 107 whose open cell pore structure consists  
2   essentially of pores whose planar cross-sectional images are of substantially the same  
3   geometric shape.

1           110.   A carbon foam as defined in claim 109 wherein said cross-sectional images  
2   are substantially circular or elliptical.

1           111.   The process of claim 100 wherein the pressure in steps (2) and (3) is  
2   substantially identical.



FIG. 1

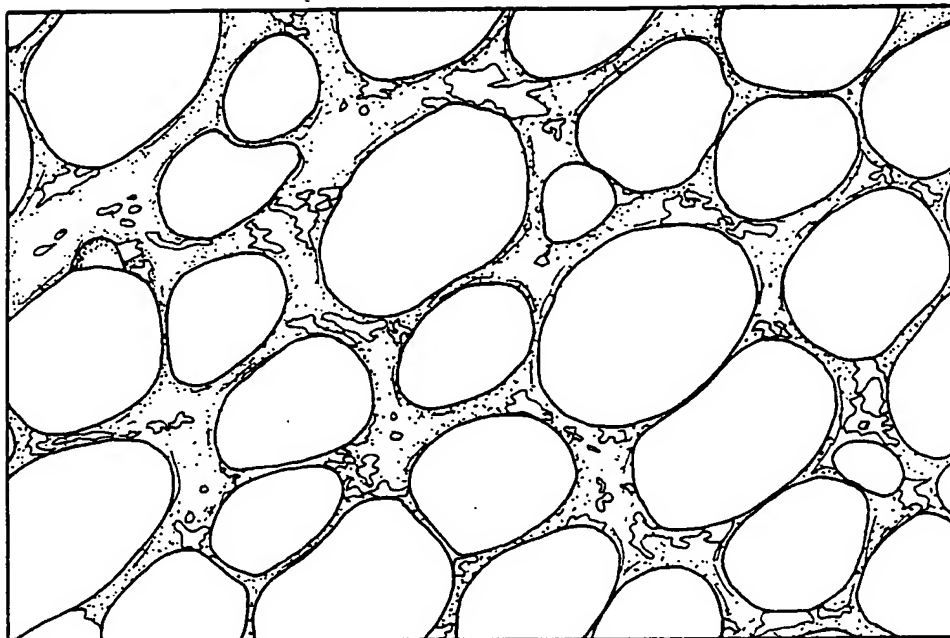


FIG. 2

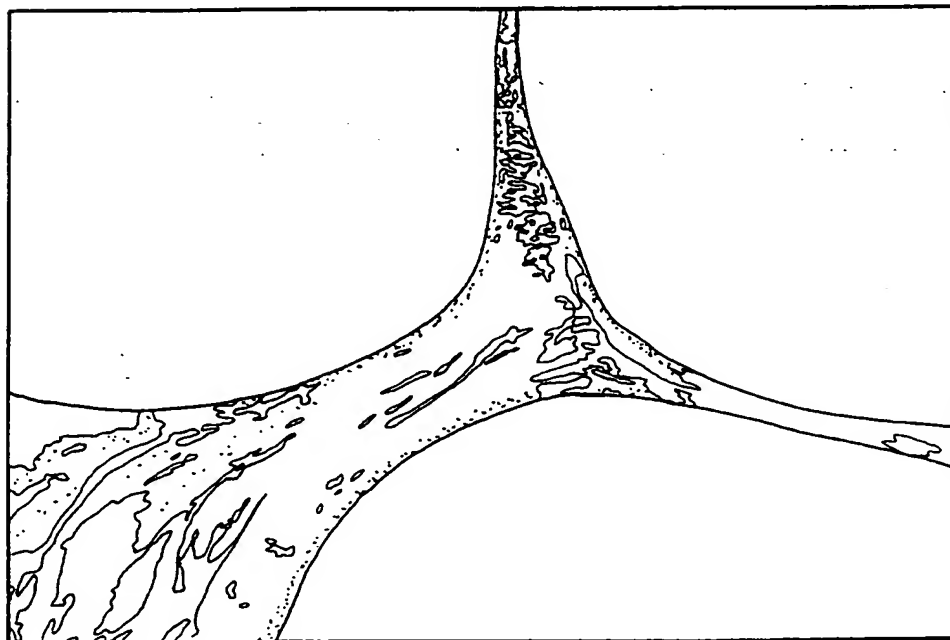


FIG. 3

3/23

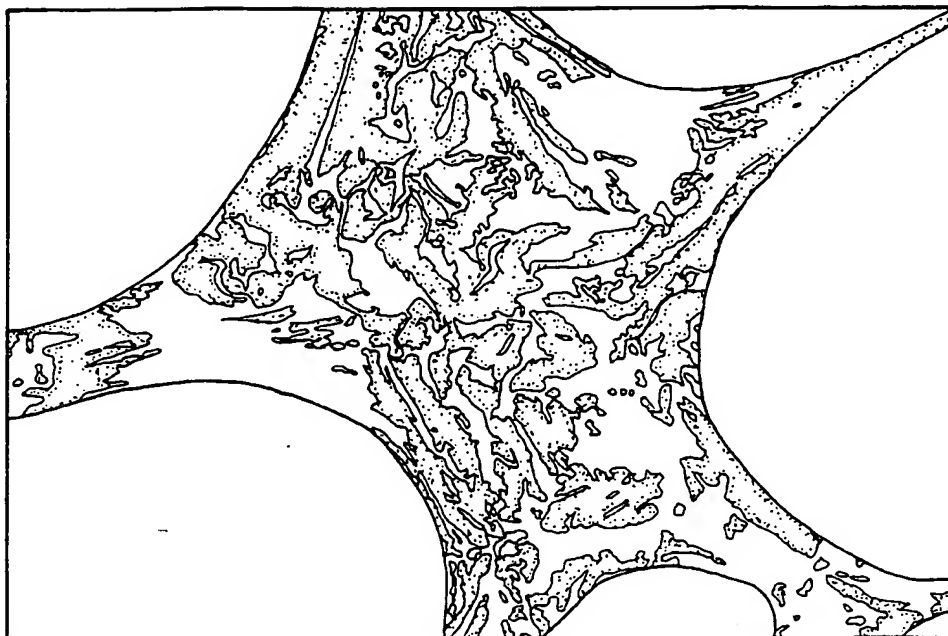


FIG. 4

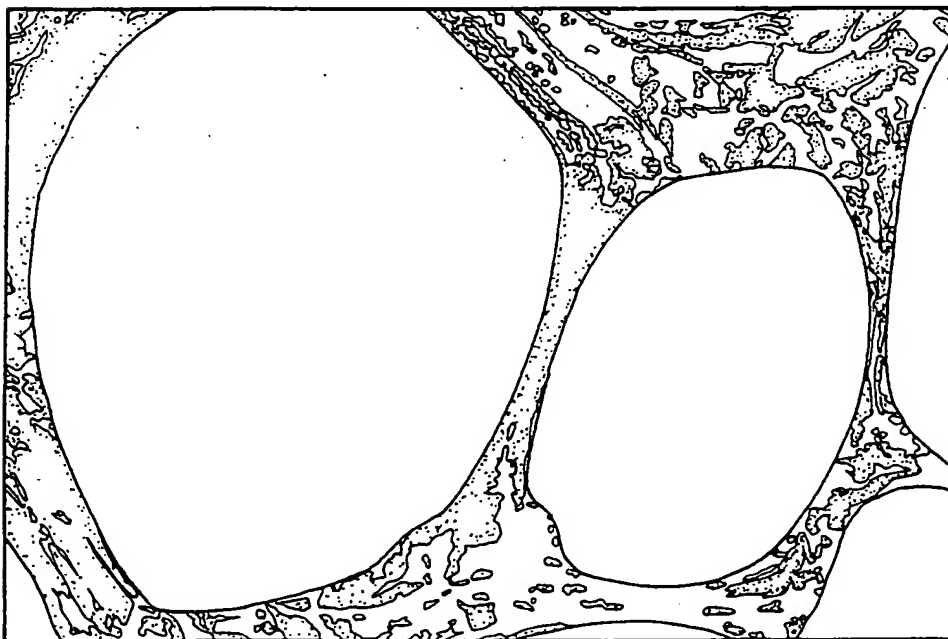


FIG. 5

4/23

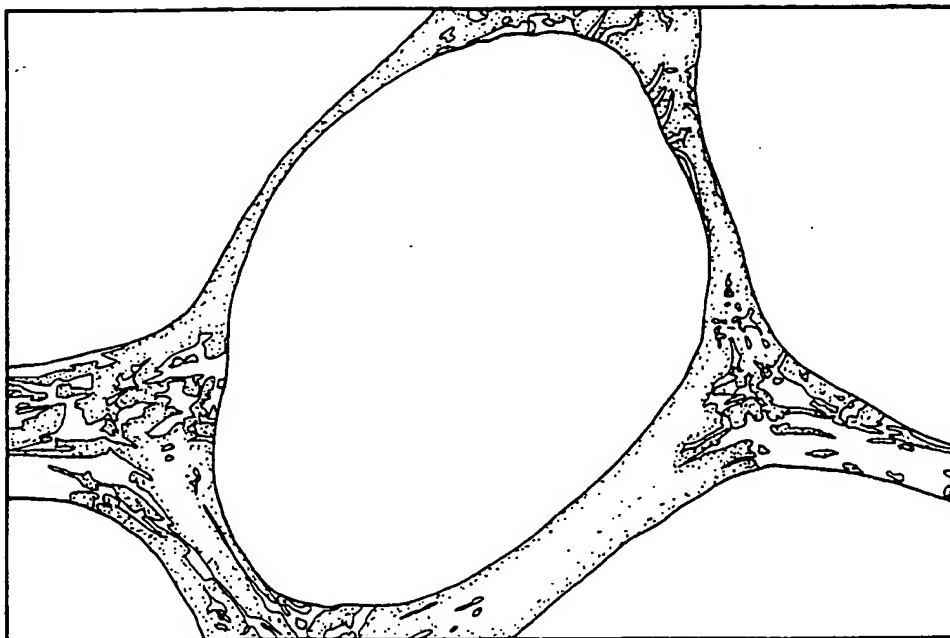


FIG. 6

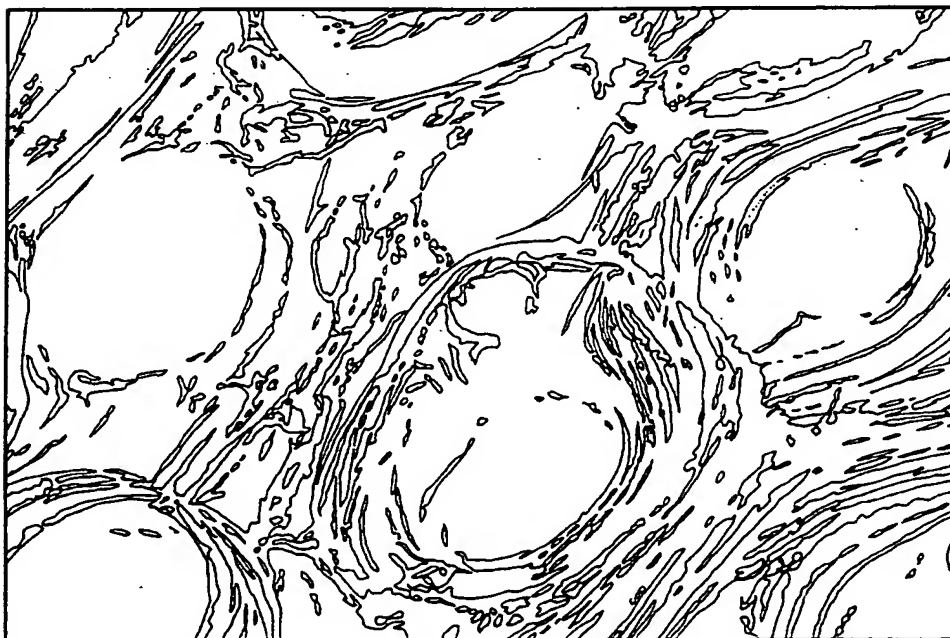
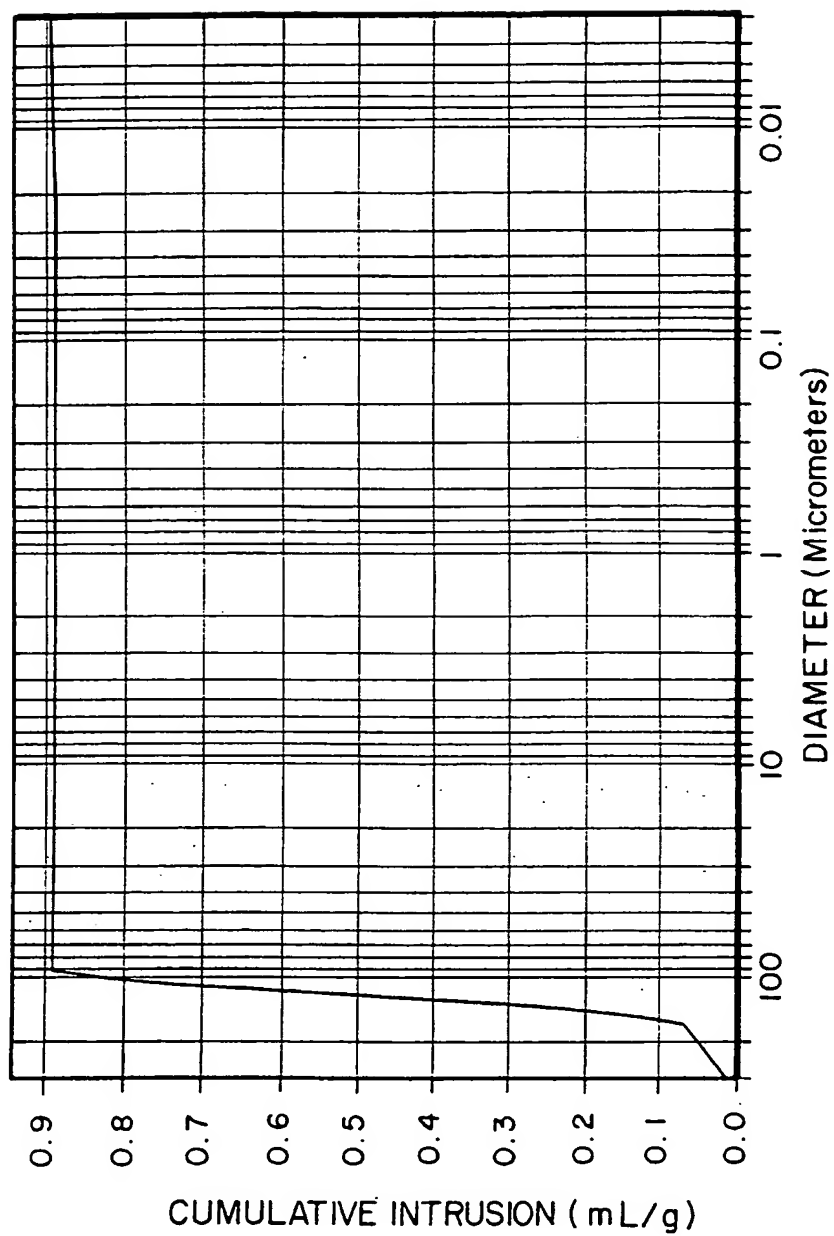


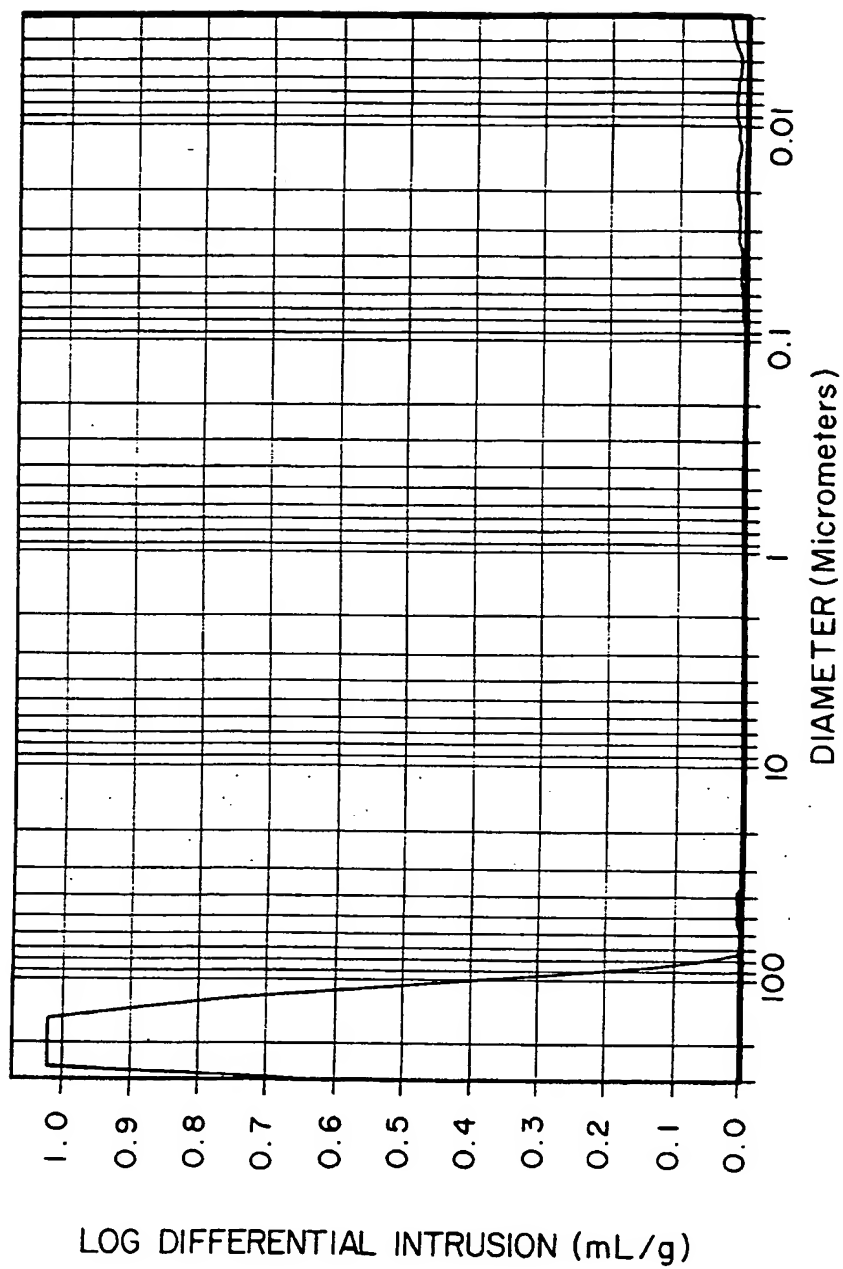
FIG. 7

5/23

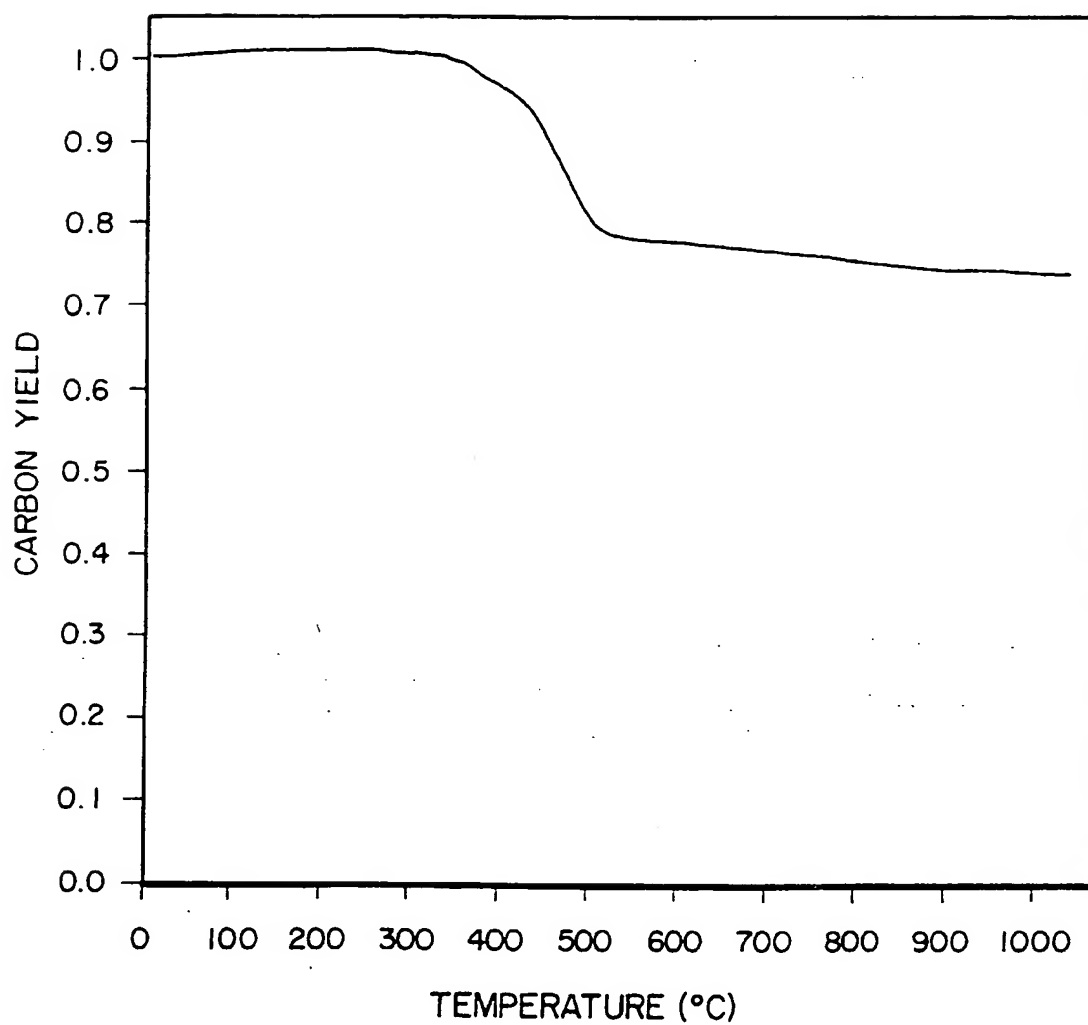
FIG. 8



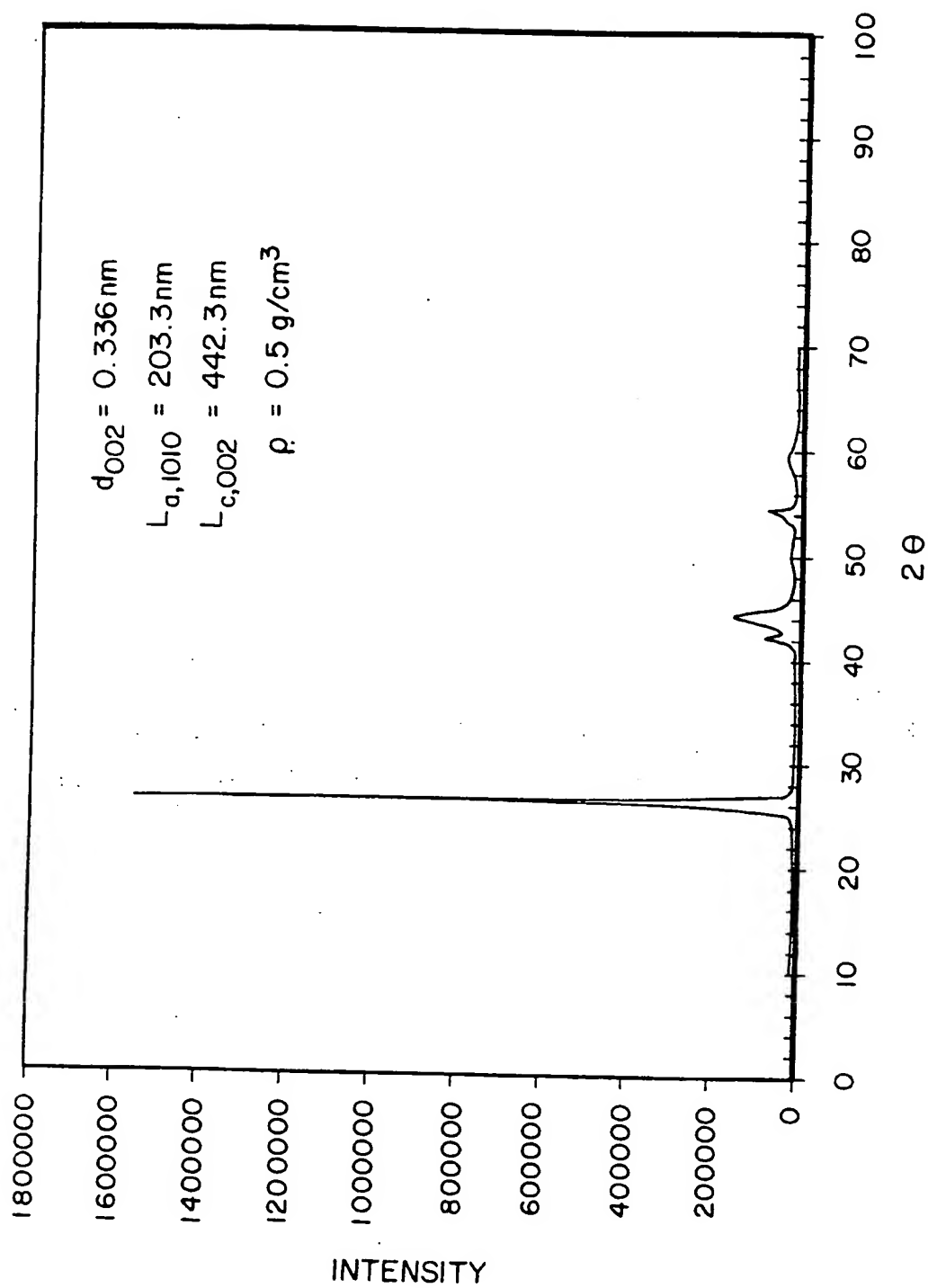
6/23

FIG. 9

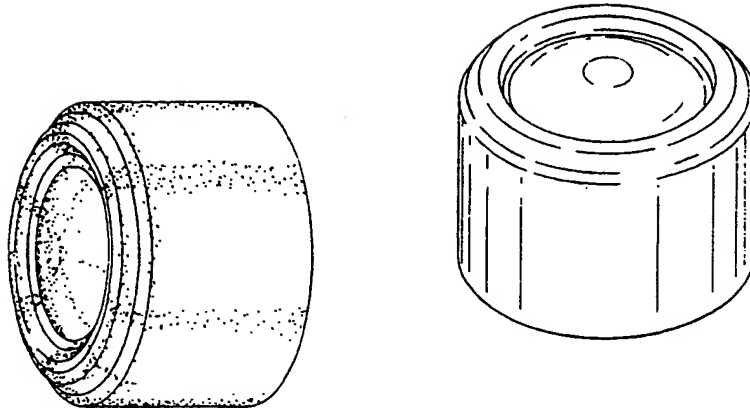
7/23

FIG. 10

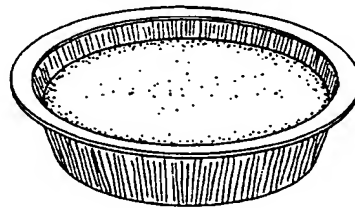
8/23

FIG. 11

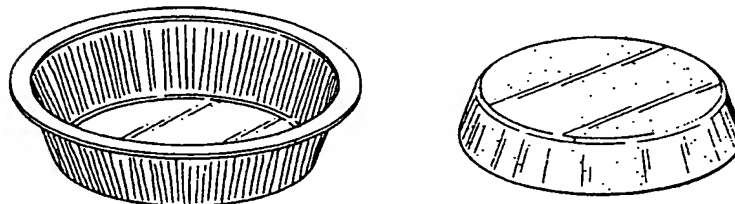
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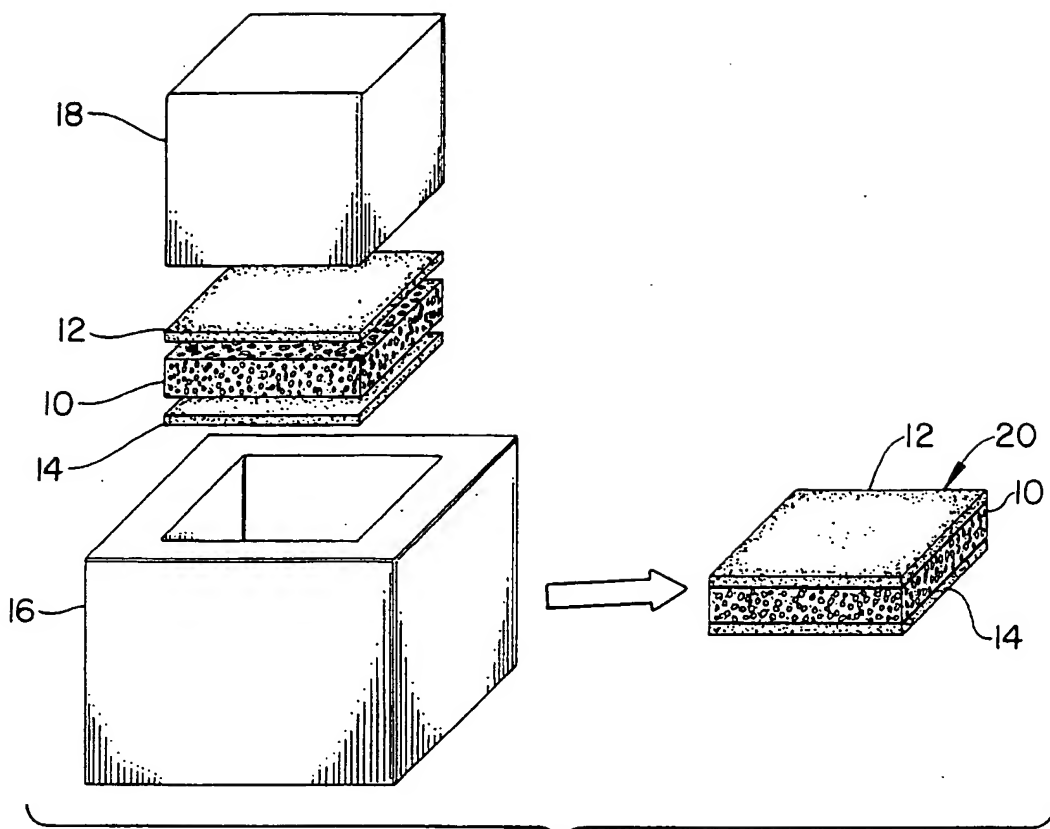
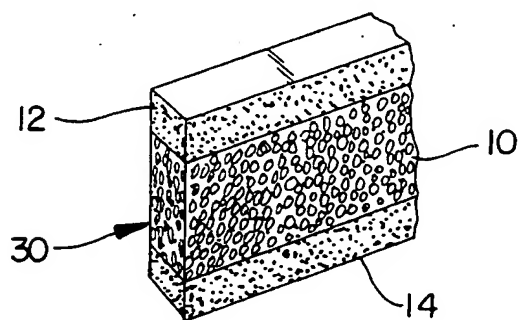
FOAM PRODUCED IN BOTTOM OF SODA CAN.

FIG. 12A

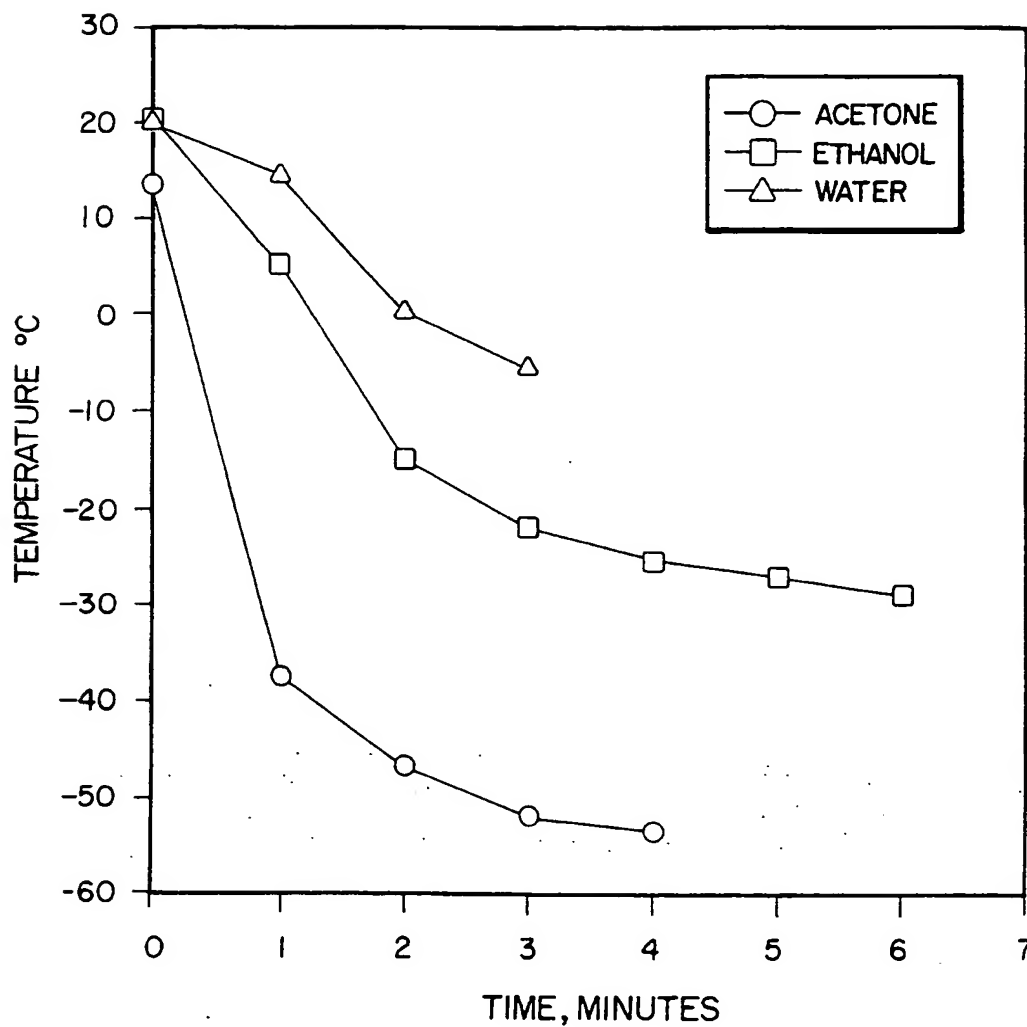
FOAM PRODUCED IN ALUMINUM WEIGHING DISH.

FIG. 12BFOAM REMOVED FROM ALUMINUM WEIGHING DISH AND  
TURNED OVER, EXPOSING SMOOTH INTEGRATED SURFACE (FACE)FIG. 12C

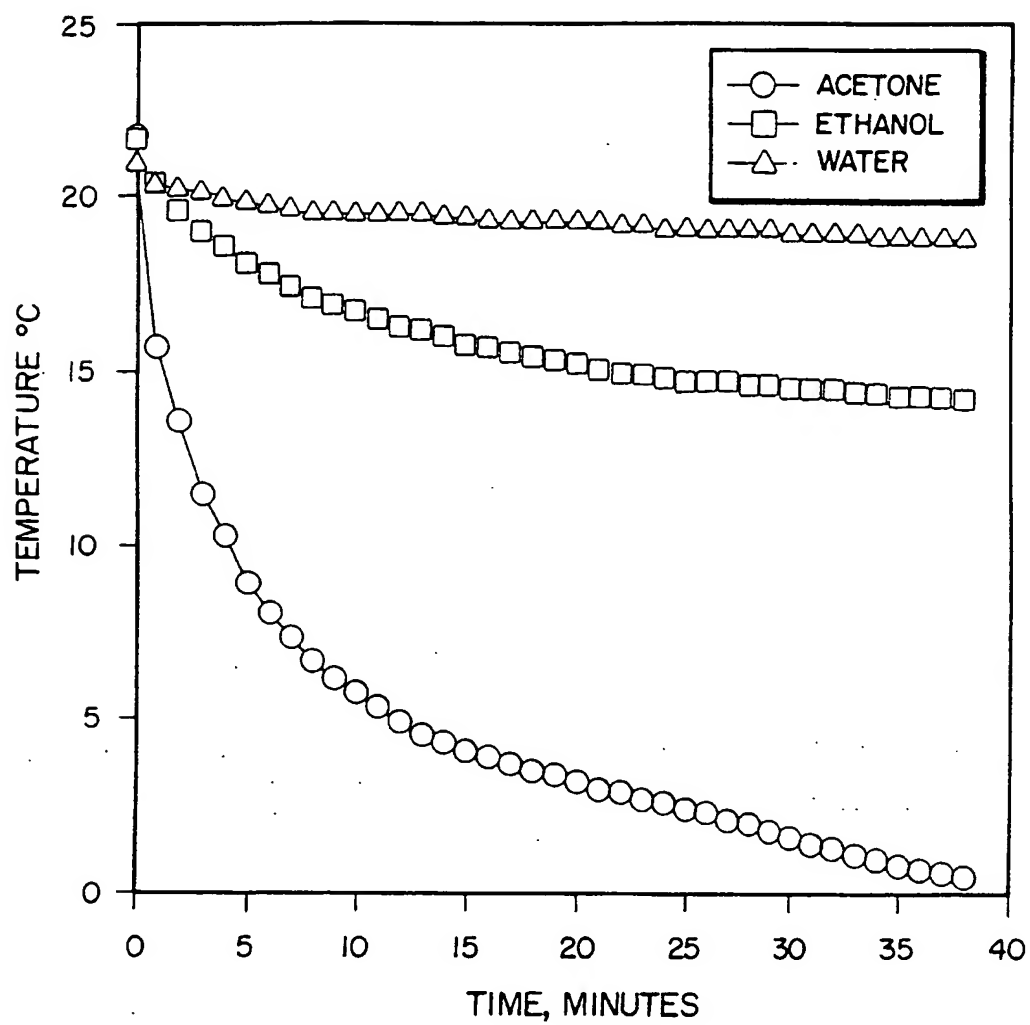
10/23

FIG. 13AFIG. 13B

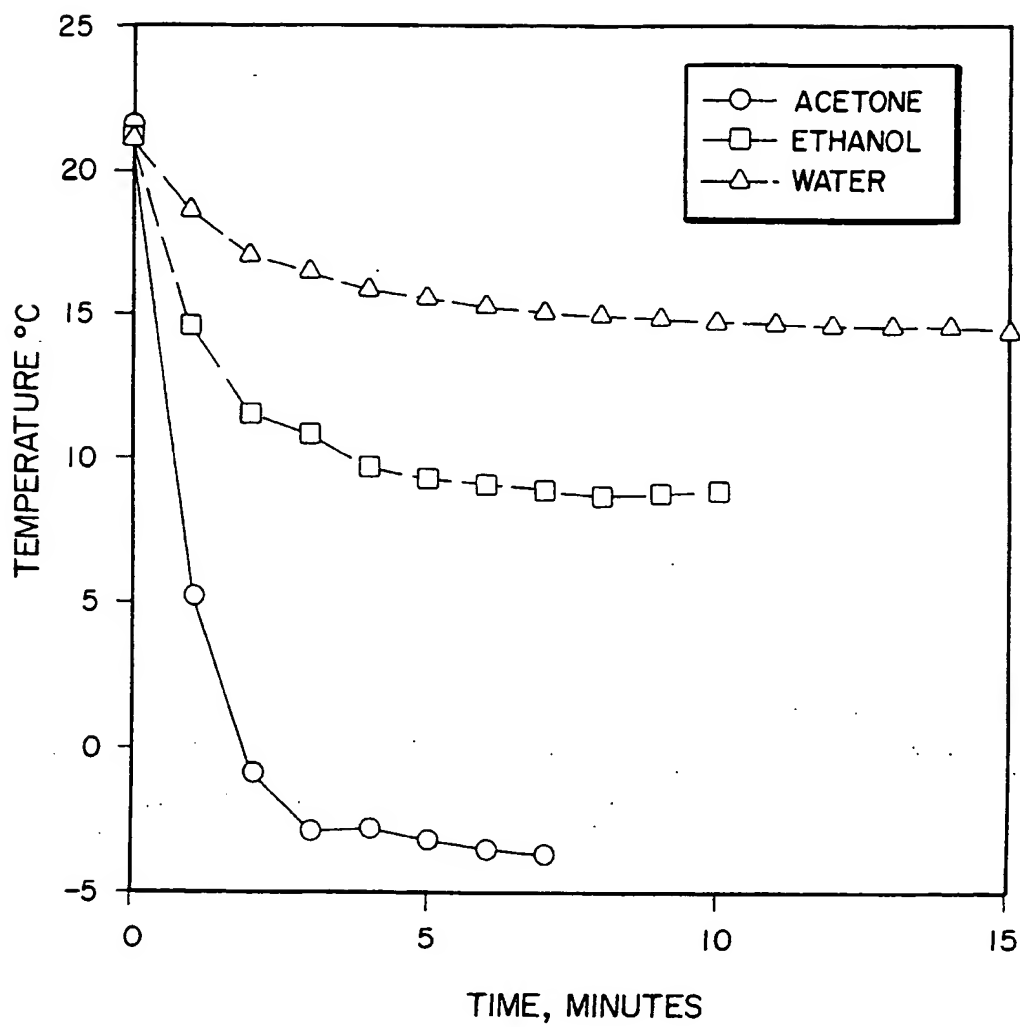
11/23

FIG. 14

12/23

FIG. 15

13/23

FIG. 16



14/23

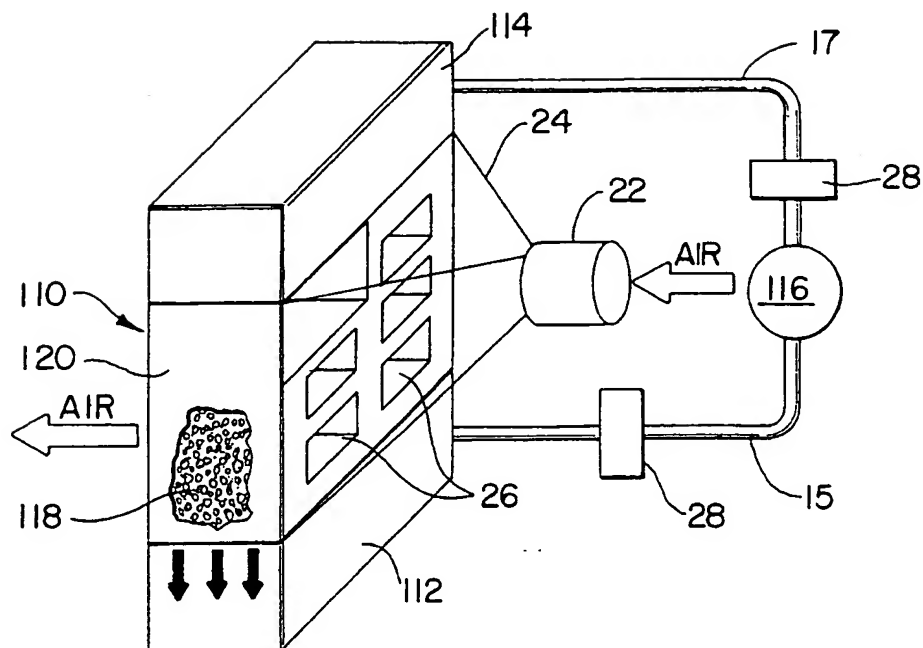


FIG. 17

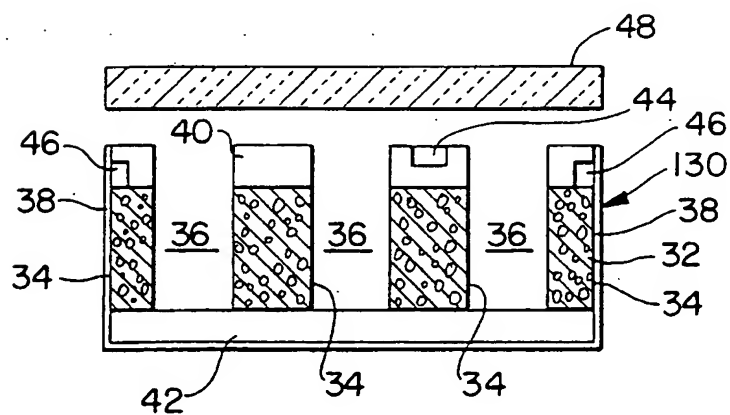
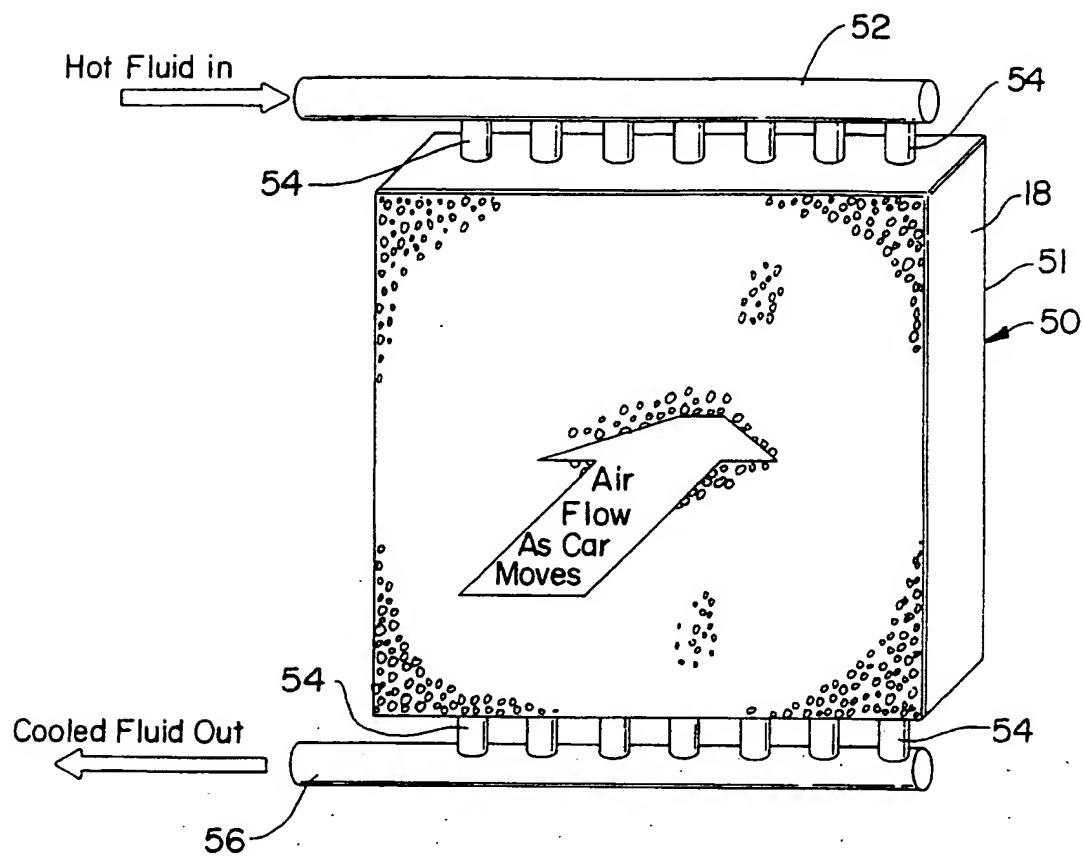


FIG. 18

15/23

FIG. 19

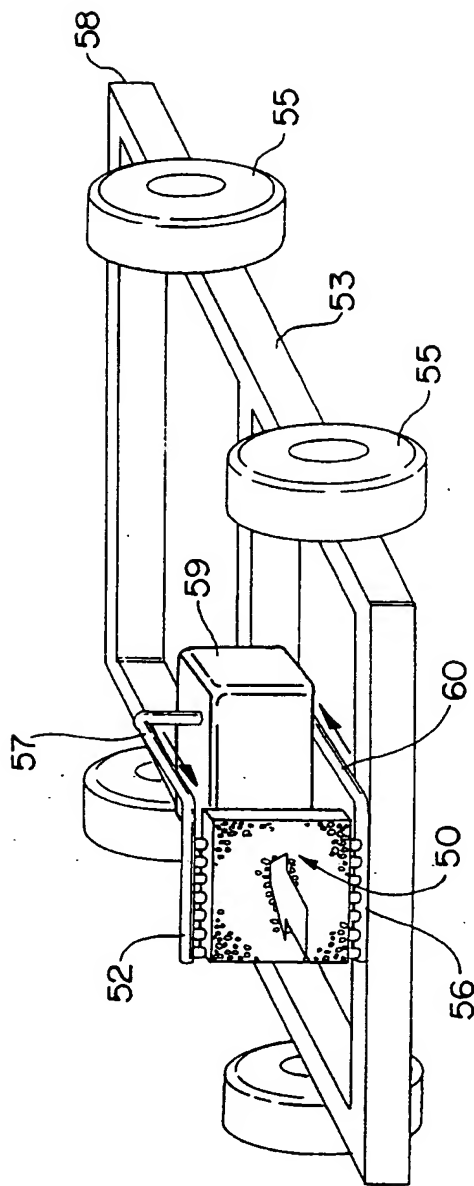
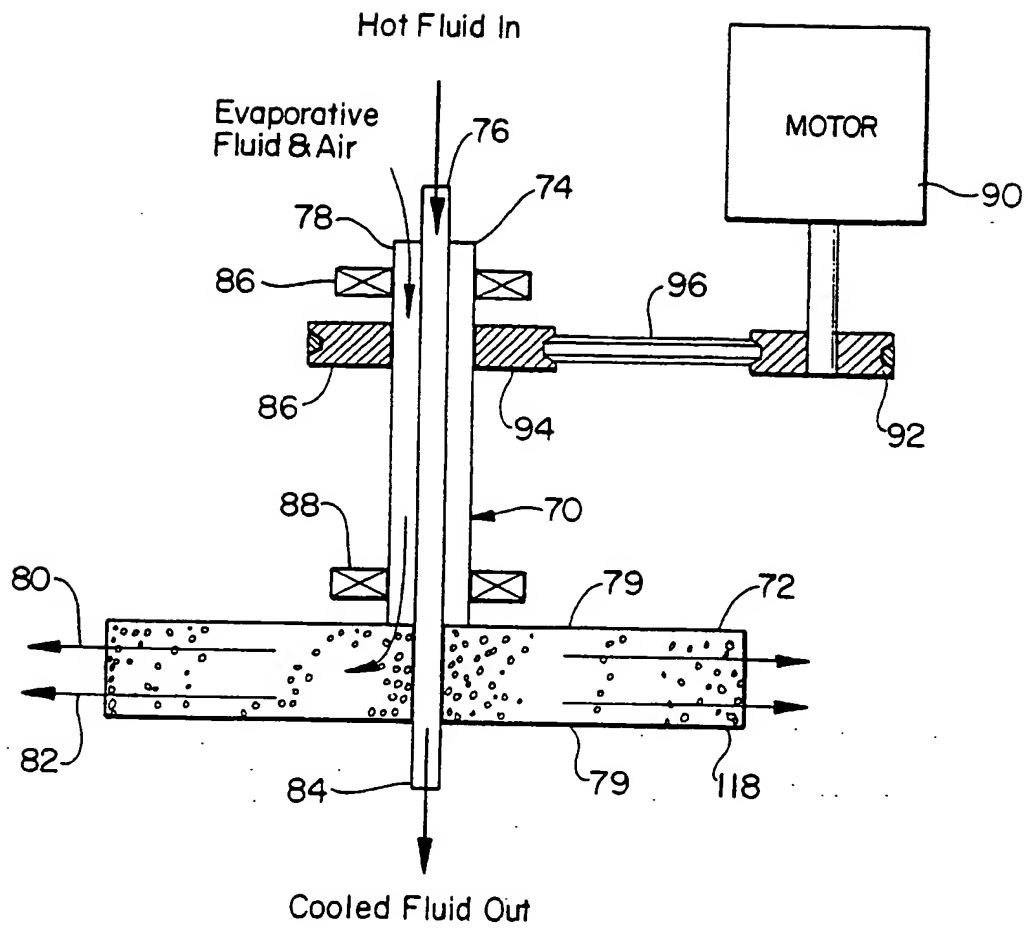


FIG. 20

17/23

FIG. 21

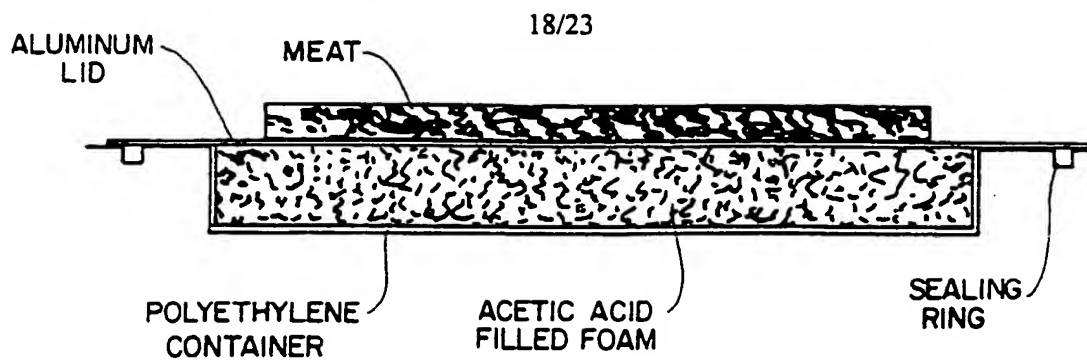


FIG. 22

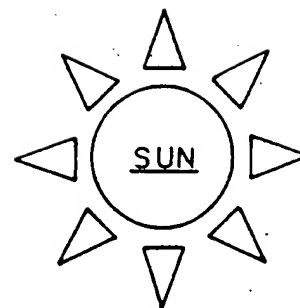
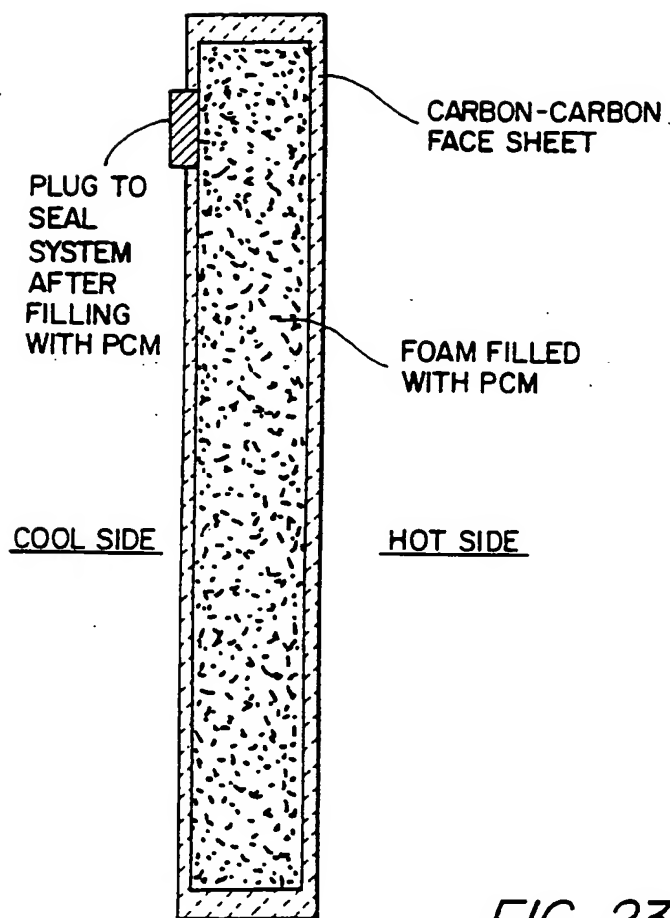
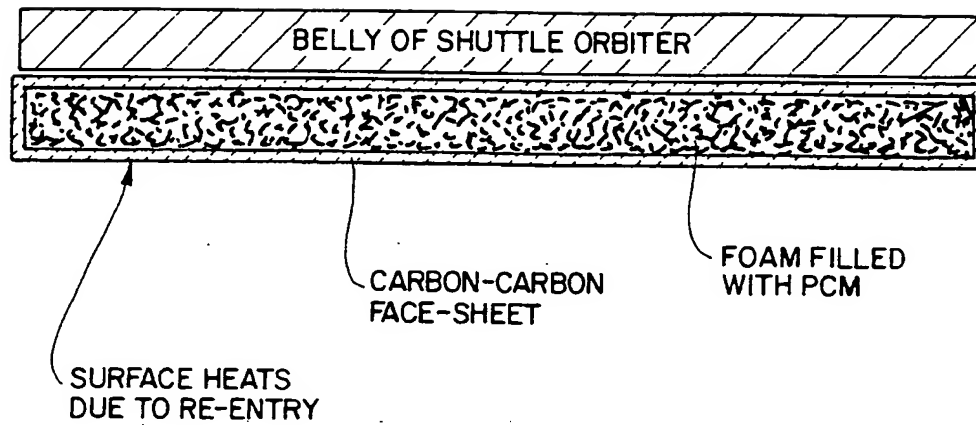


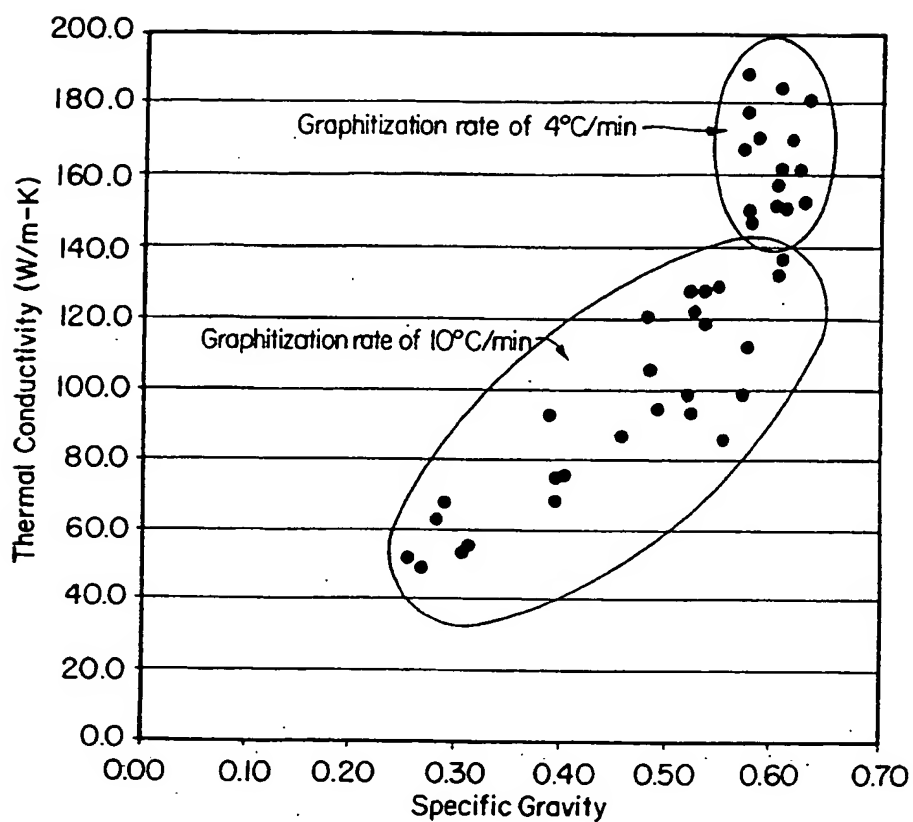
FIG. 23

19/23

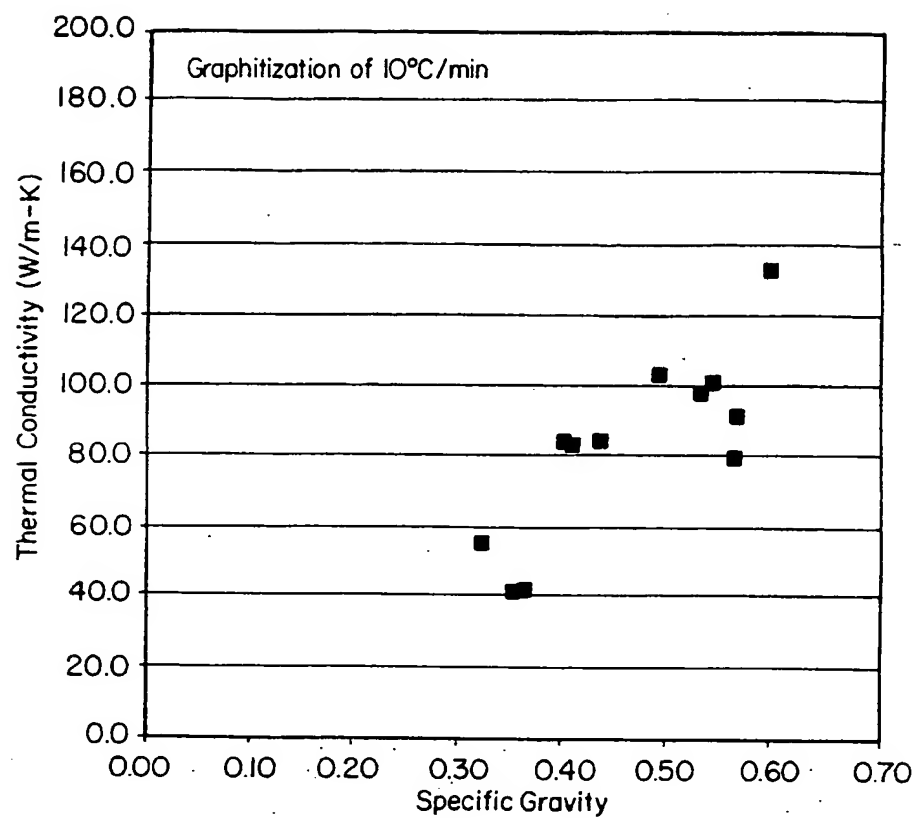


PCM KEEPS SURFACE COOLER  
THAN WITH NORMAL MATERIALS

FIG. 24

FIG. 25

21/23

FIG. 26



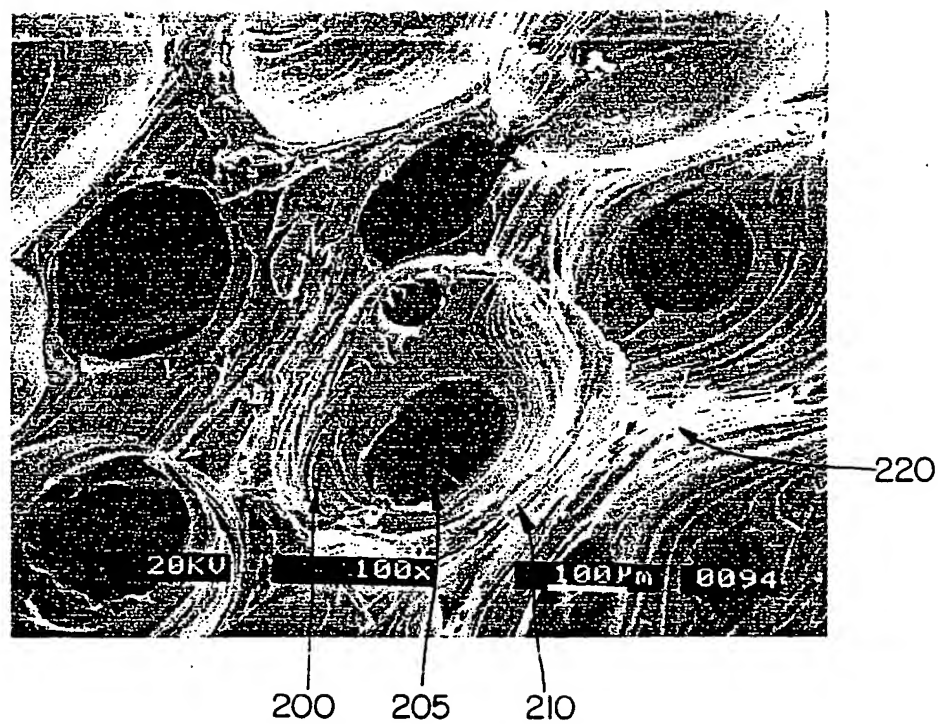


FIG. 27

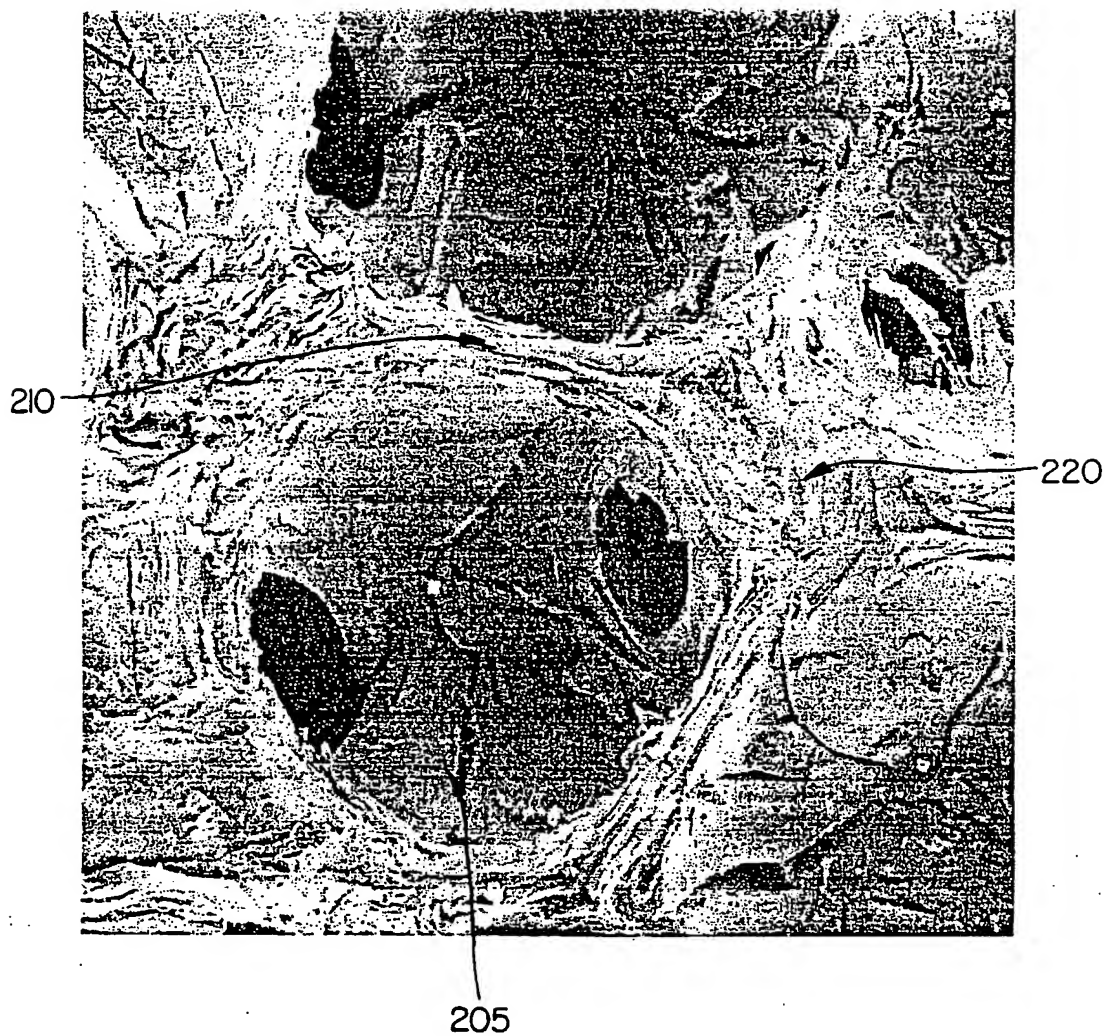


FIG. 28

(19) World Intellectual Property Organization  
International Bureau



(43) International Publication Date  
13 September 2001 (13.09.2001)

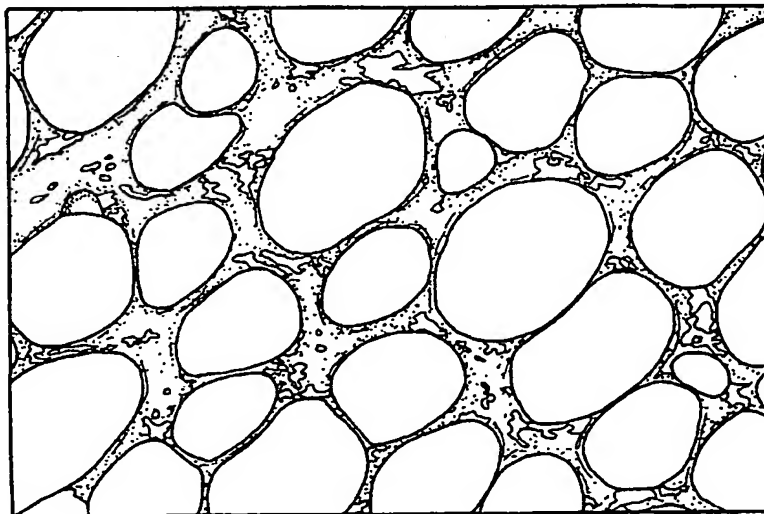
PCT

(10) International Publication Number  
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09/519,559 6 March 2000 (06.03.2000) US
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- (72) Inventors: KLETT, James, W.; 10073 Delle Meade Drive, Knoxville, TN 37931 (US). BURCHELL, Timothy, D.; 109 Greywood Place, Oak Ridge, TN 37830 (US). CHOUDHURY, Ashok; 111 Mohawk Road, Oak Ridge, TN 37830 (US).
- (74) Agents: NELSON, Gregory, A. et al.; Akerman, Senterfitt & Eidson, PA, Suite 400, 222 Lakeview Avenue, P.O. Box 3188, West Palm Beach, FL 33402-3188 (US).
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- (84) Designated States (*regional*): ARIPO patent (GH, GM, KE, LS, MW, MZ, SD, SL, SZ, TZ, UG, ZW), Eurasian patent (AM, AZ, BY, KG, KZ, MD, RU, TJ, TM), European patent (AT, BE, CH, CY, DE, DK, ES, FI, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE, TR), OAPI patent (BF, BJ, CF, CG, CI, CM, GA, GN, GW, ML, MR, NE, SN, TD, TG).
- Published:  
— with international search report

[Continued on next page]

(54) Title: PITCH-BASED CARBON FOAM AND COMPOSITES AND USES THEREOF



(57) Abstract: A thermally conductive carbon foam is provided, normally having a thermal conductivity of at least 40 W/m.K. The carbon foam usually has a specific thermal conductivity, defined as the thermal conductivity divided by the density, of at least about 75 W.cm<sup>3</sup>/m.<sup>3</sup>K.gm. The foam also has a high specific surface area, typically at least about 6,000 m<sup>2</sup>/m<sup>3</sup>. The foam is characterized by an x-ray diffraction pattern having "doublet" 100 and 101 peaks characterized by a relative peak split factor no greater than about 0.470. The foam is graphitic and exhibits substantially isotropic thermal conductivity. The foam comprises substantially ellipsoidal pores and the mean pore diameter of such pores is preferably no greater than about 340 microns. Other materials, such as phase change materials, can be impregnated in the pores in order to impart beneficial thermal properties to the foam. Heat exchange devices and evaporatively cooled heat sinks utilizing the foams are also disclosed.

WO 01/66490 A3



(88) Date of publication of the international search report:  
7 March 2002

*For two-letter codes and other abbreviations, refer to the "Guidance Notes on Codes and Abbreviations" appearing at the beginning of each regular issue of the PCT Gazette.*

# INTERNATIONAL SEARCH REPORT

International Application No

PCT/US 01/07012

## A. CLASSIFICATION OF SUBJECT MATTER

IPC 7 C04B35/52 C04B38/00 F28F21/02 F28C3/08 C04B40/00

According to International Patent Classification (IPC) or to both national classification and IPC

## B. FIELDS SEARCHED

Minimum documentation searched (classification system followed by classification symbols)

IPC 7 C04B F28F F28C

Documentation searched other than minimum documentation to the extent that such documents are included in the fields searched

Electronic data base consulted during the international search (name of data base and, where practical, search terms used)

EPO-Internal, WPI Data, PAJ

## C. DOCUMENTS CONSIDERED TO BE RELEVANT

Category *	Citation of document, with indication, where appropriate, of the relevant passages	Relevant to claim No.
X	WO 99 11586 A (LOCKHEED MARTIN ENERGY RESEARCH CORP.) 11 March 1999 (1999-03-11) cited in the application the whole document ---	1-88, 107-111
X	WO 99 11585 A (LOCKHEED MARTIN ENERGY RESEARCH CORP.) 11 March 1999 (1999-03-11) cited in the application the whole document ---	1-61, 64-111
X	WO 99 64223 A (LOCKHEED MARTIN ENERGY RESEARCH CORP.) 16 December 1999 (1999-12-16) cited in the application the whole document -----	1-61, 63-111

☐ Further documents are listed in the continuation of box C.

☒ Patent family members are listed in annex.

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Date of the actual completion of the international search

12 September 2001

Date of mailing of the international search report

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Fax: (+31-70) 340-3016

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Hauck, H

# INTERNATIONAL SEARCH REPORT

Information on patent family members

International Application No

PCT/US 01/07012

Patent document cited in search report	Publication date	Patent family member(s)	Publication date
WO 9911586 A	11-03-1999	AU 9294998 A US 6037032 A	22-03-1999 14-03-2000
WO 9911585 A	11-03-1999	US 6033506 A AU 8687998 A BR 9811430 A EP 1017649 A US 6037032 A US 6261485 B	07-03-2000 22-03-1999 22-08-2000 12-07-2000 14-03-2000 17-07-2001
WO 9964223 A	16-12-1999	US 6037032 A AU 3742699 A BR 9911000 A	14-03-2000 30-12-1999 20-02-2001